

Temperature offsets in AMDAR and other airborne measurements

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Motivation

→ Systematic errors

- measurements depend on flight dynamic state

→ Example:

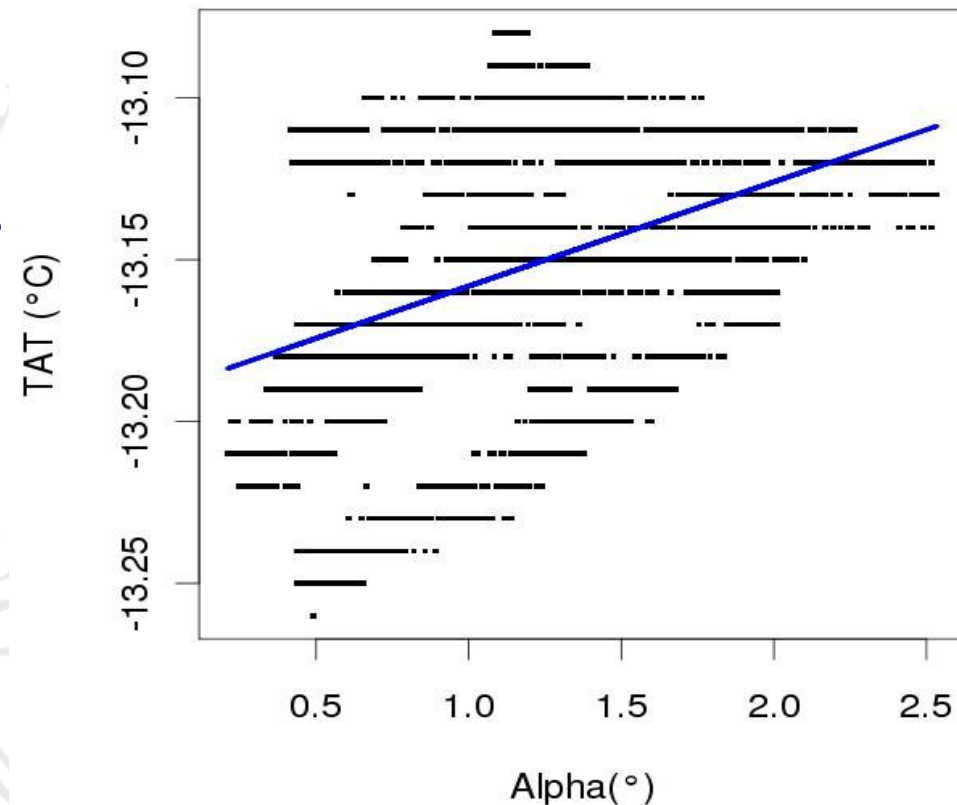
- Dornier 228
- TAT vs. angle of attack: 0.05 K/°

→ Most maneuvering during takeoff & approach

- here E-AMDAR takes profiles



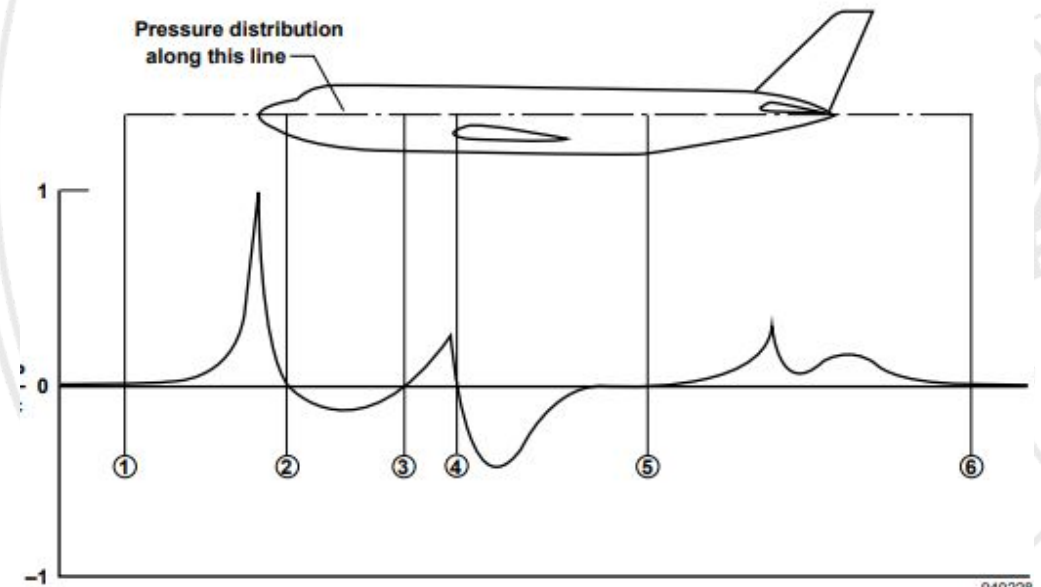
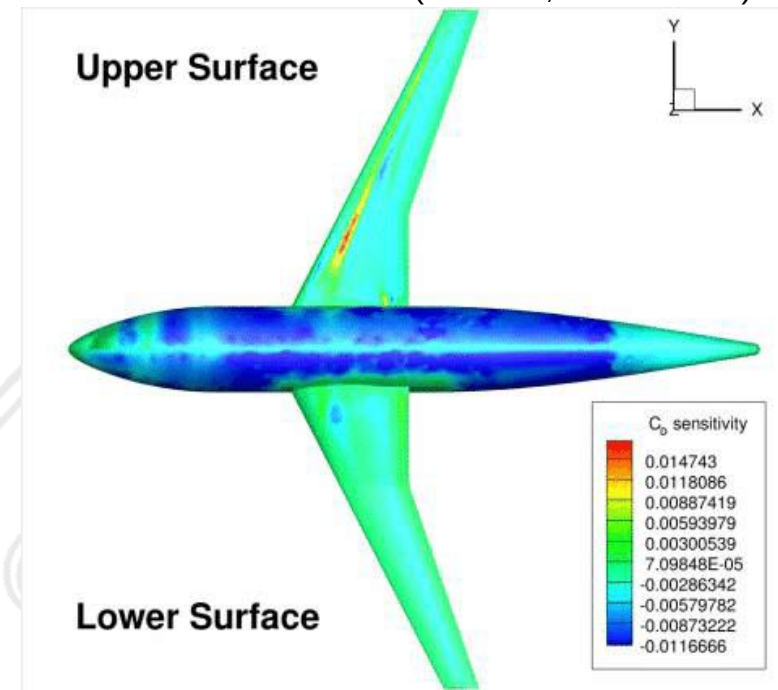
Clemens Drüe, IMUK Hannover



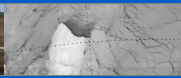
Highly structured flow

- Flow field around aircraft
- Complex flow field
 - flight speed below but same order as sound speed
 - Bernoulli effect, standing waves, harmonics.
- Depending on
 - airspeed
 - loading
 - (wing) configuration
- $p = f(x, y, z, v, \dots)$

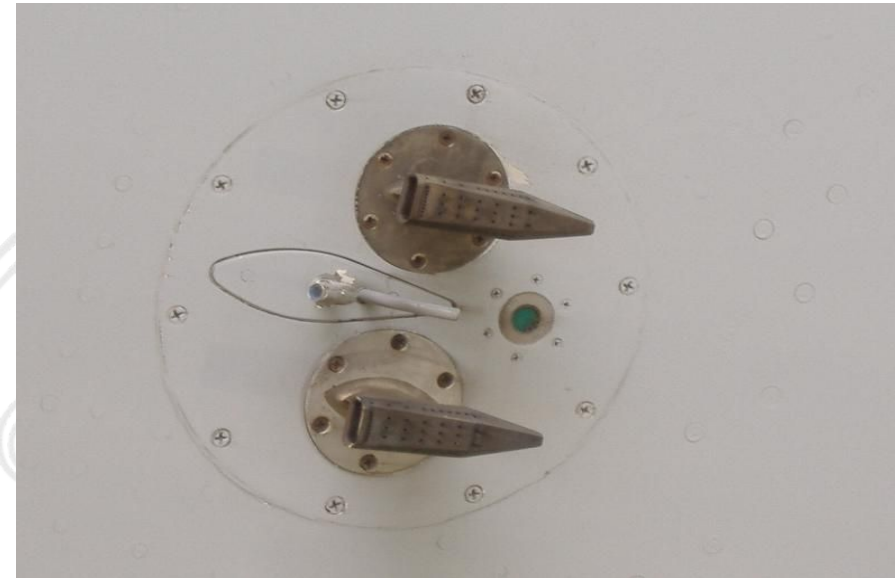
(Palacios, F et al. 2014)



(Palacios, F et al. 2014)



Thermometer housings (Rosemount)



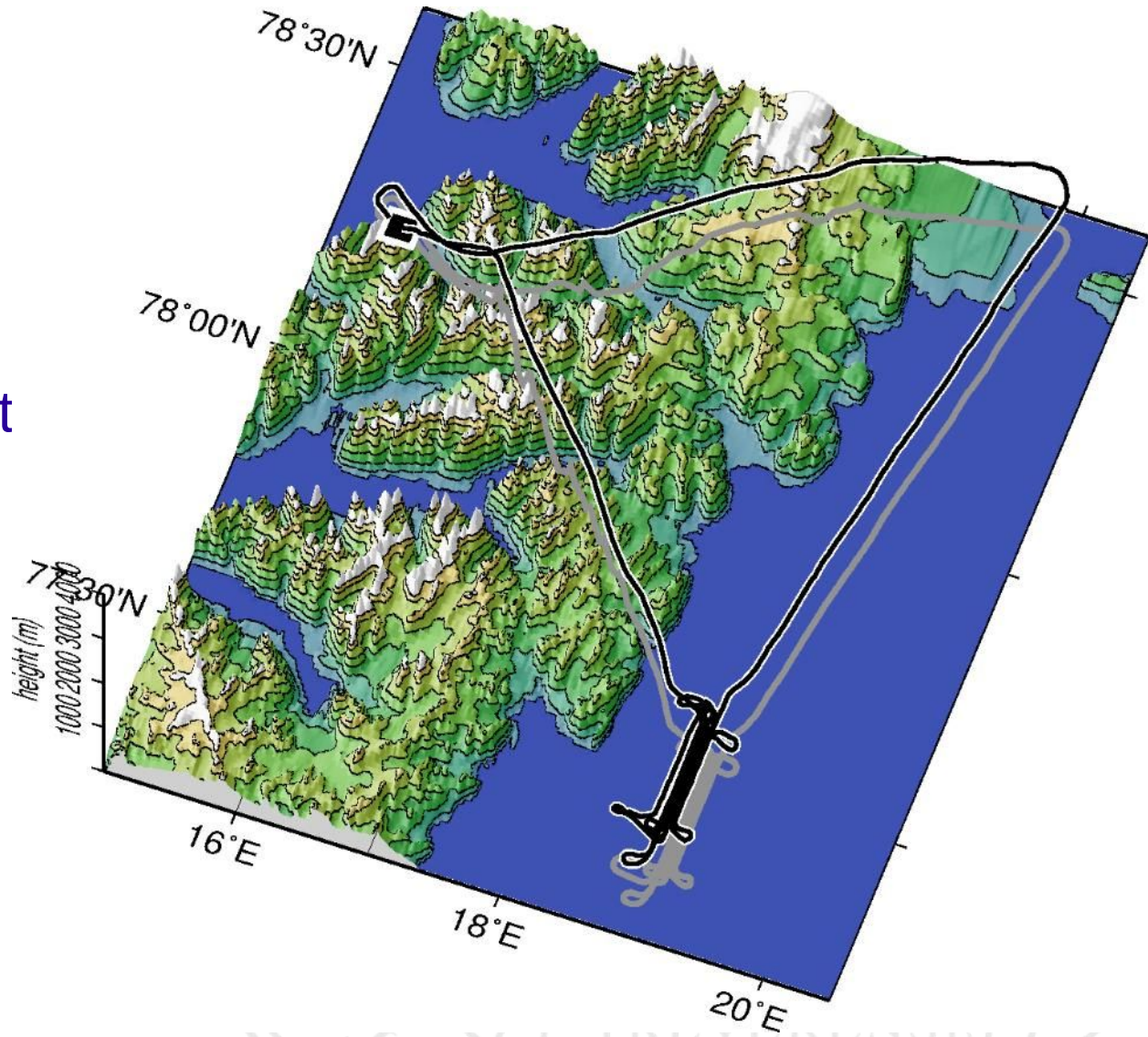
Solution in-flight calibration

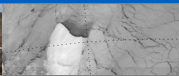
→ Research aircraft:

- custom routine
- lowest effort, maximum accuracy
- only for measurement configuration

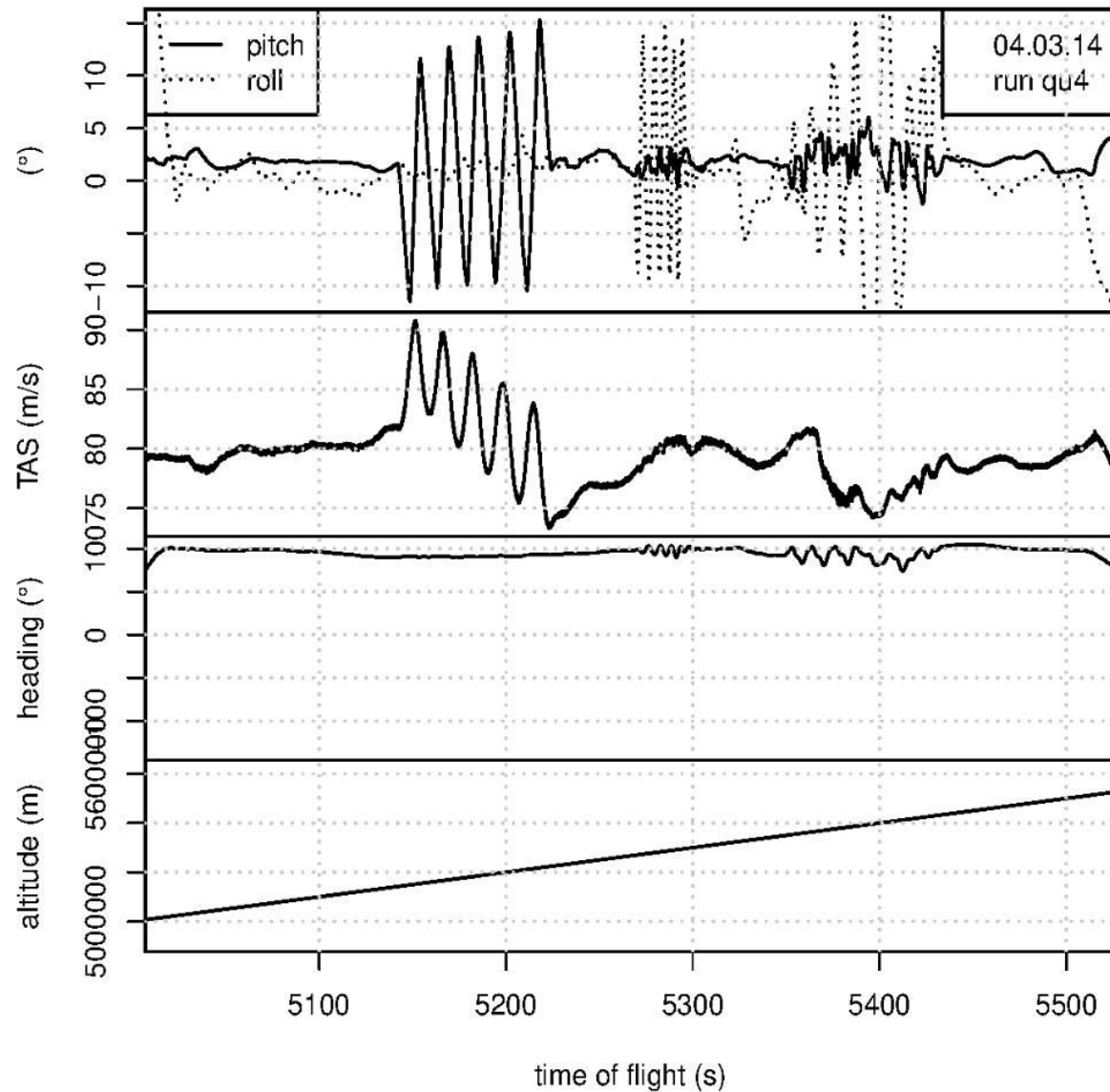
→ Commercial aircraft

- regulatory procedure
- costly process, maximum reliability
- for all nominal maneuvers





Data example (LEAST experiment 2014)



Initial Error study: Data Set EDDF

Dataset by DLR

AMDAR data acquired by DLR
from Lufthansa

22 days in 2004

Descents to **Frankfurt (FRA)**

High-resolution,
high-frequency data

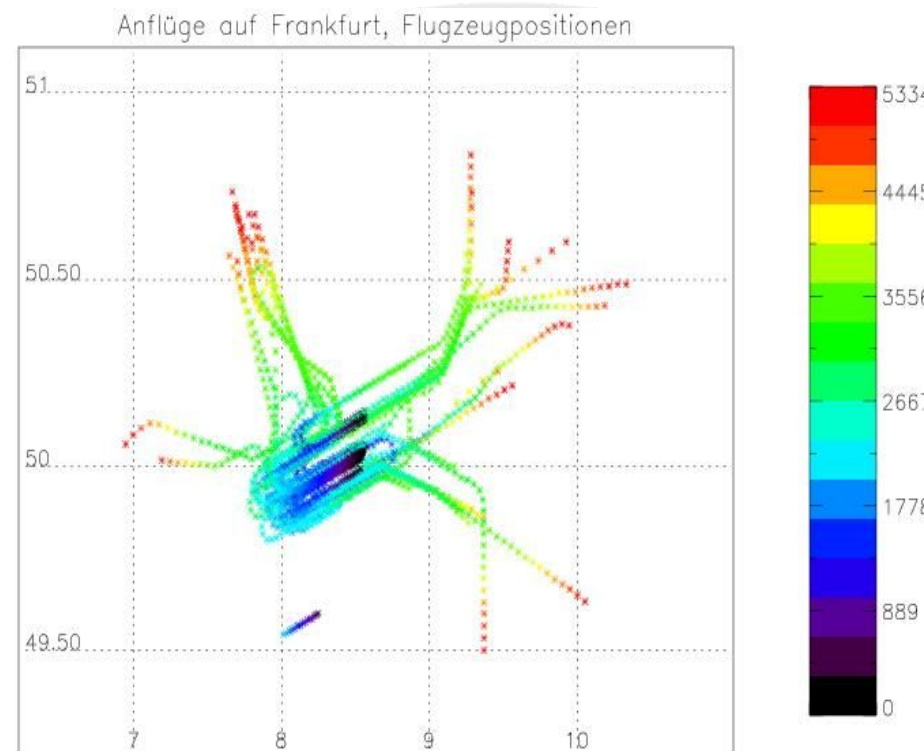
Same airline / same software

Aircraft-type dependent

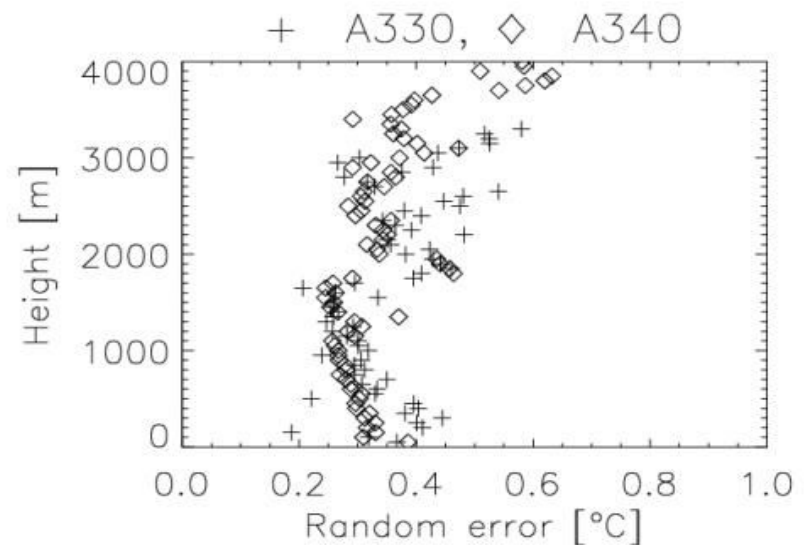
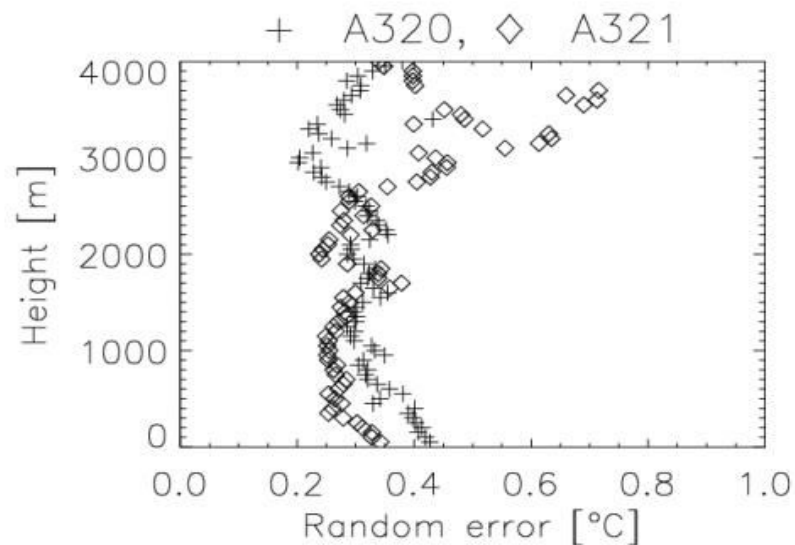
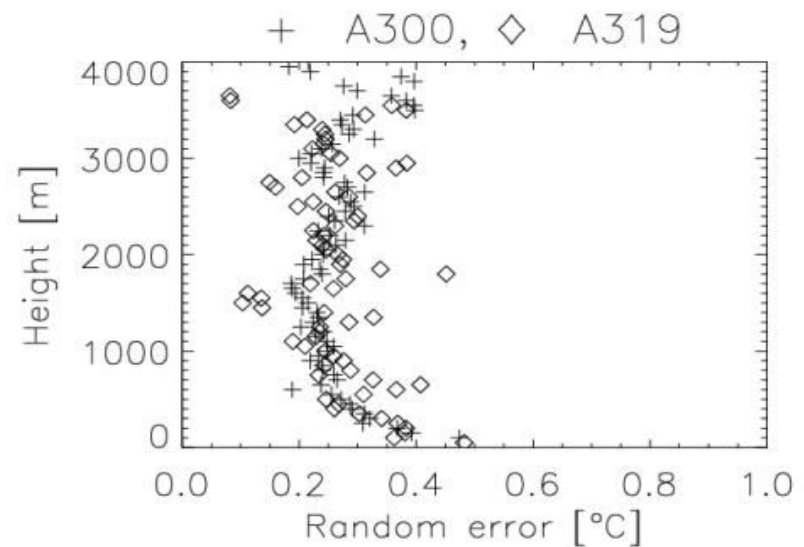
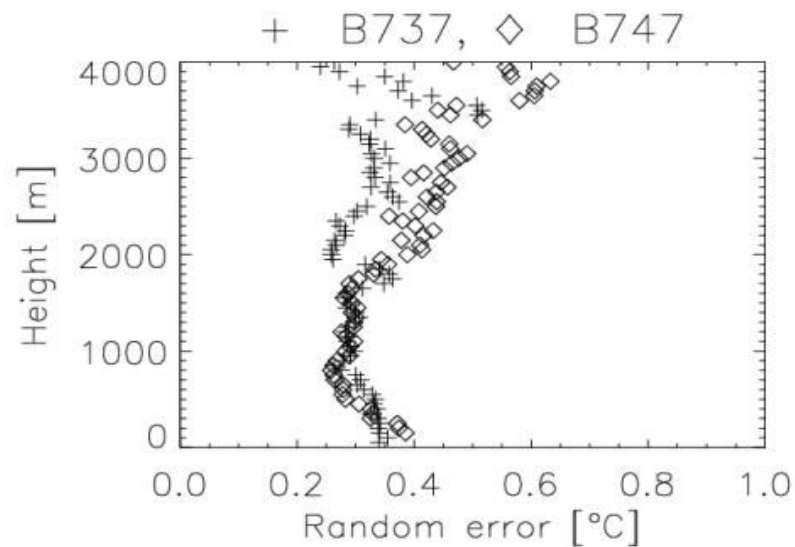
bias

root-mean-square (rms) error

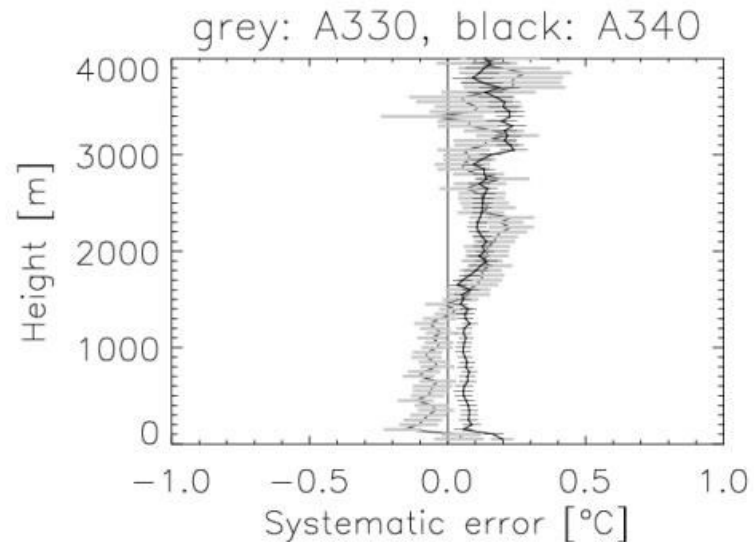
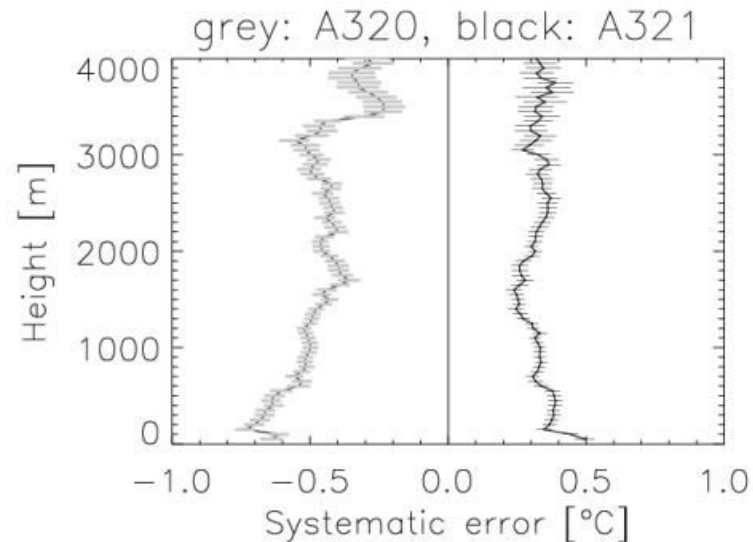
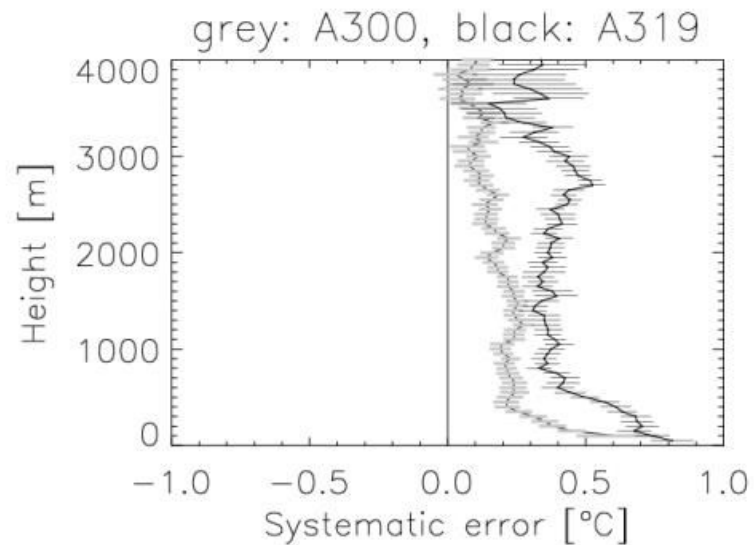
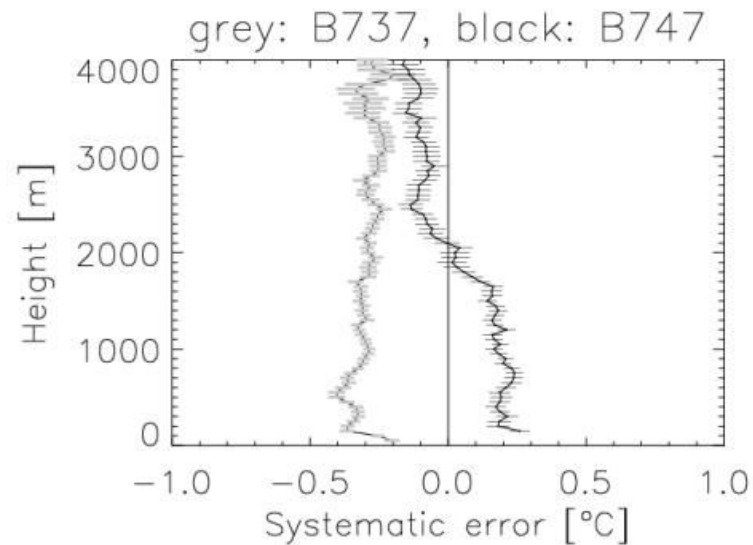
Hourly mean profile (all types) used as reference



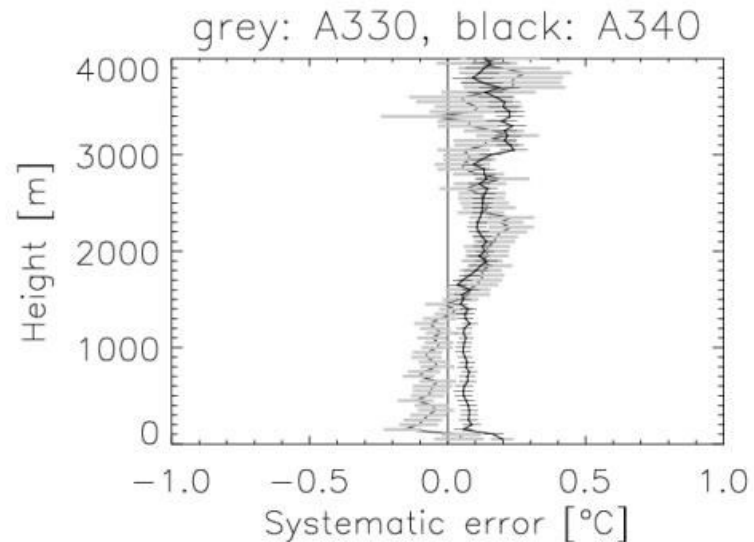
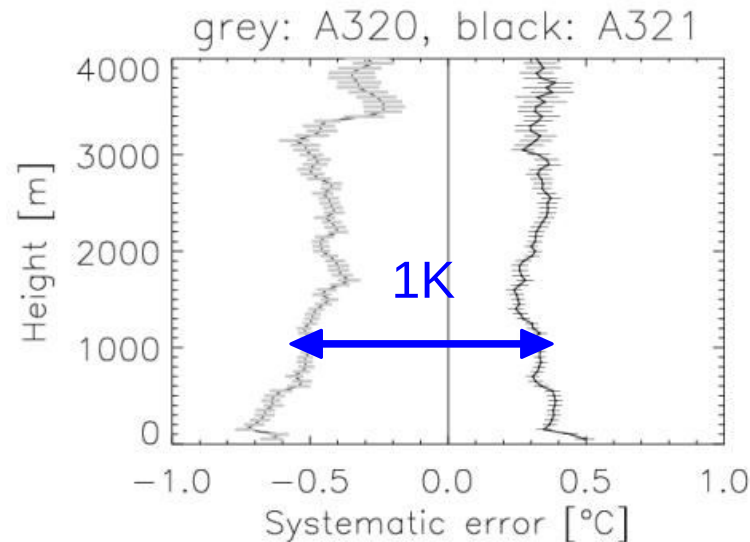
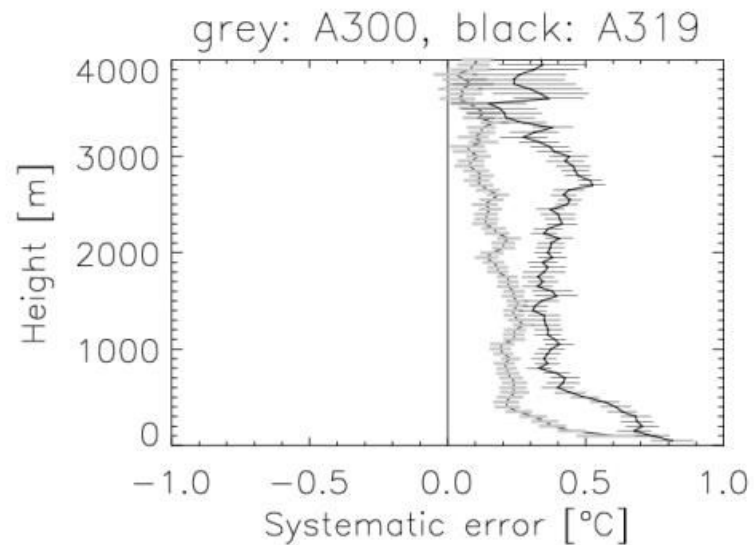
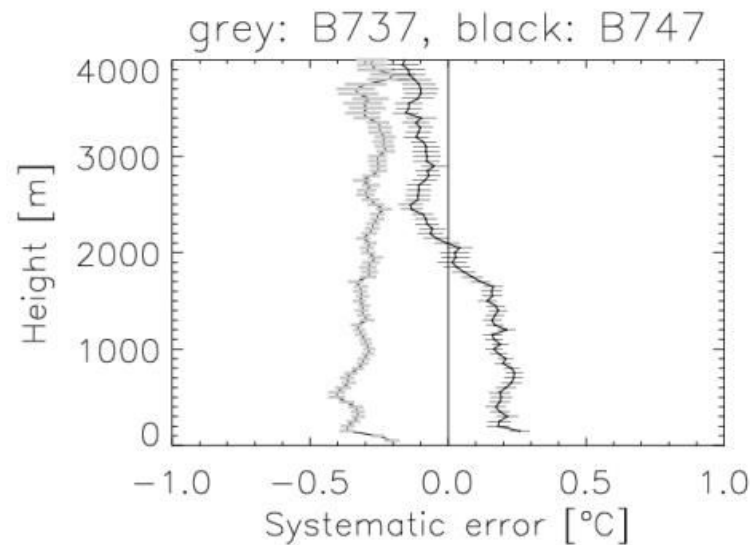
Random temperature errors



Systematic temperature errors



Error study: Systematic Temperature Errors





Error study: Temperature Errors

All rms errors around 0.3 - 0.4 K

Differing systematic bias values:

a/c type	systematic error behavior
B737	0.35 K too cold
B747	no specific trend, slight decrease with altitude
A300	0.2 K too warm
A319	0.3 - 0.4 K too warm
A320	0.5 K too cold
A321	0.3 K too warm, comparable to A319
A330	no significant trend, slight increase with altitude
A340	0.1 K too warm

Error study: Wind error vector

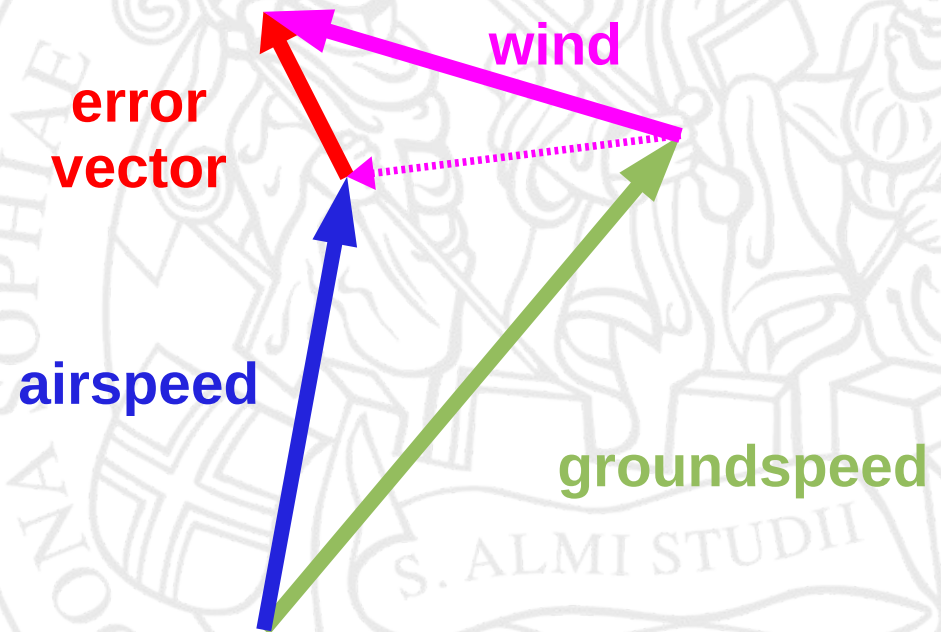
Wind speed (scalar) error seems completely random

Assumption: Aircraft-fixed error vector

Calculate errors
in wind components

longitudinal and
lateral to flight path

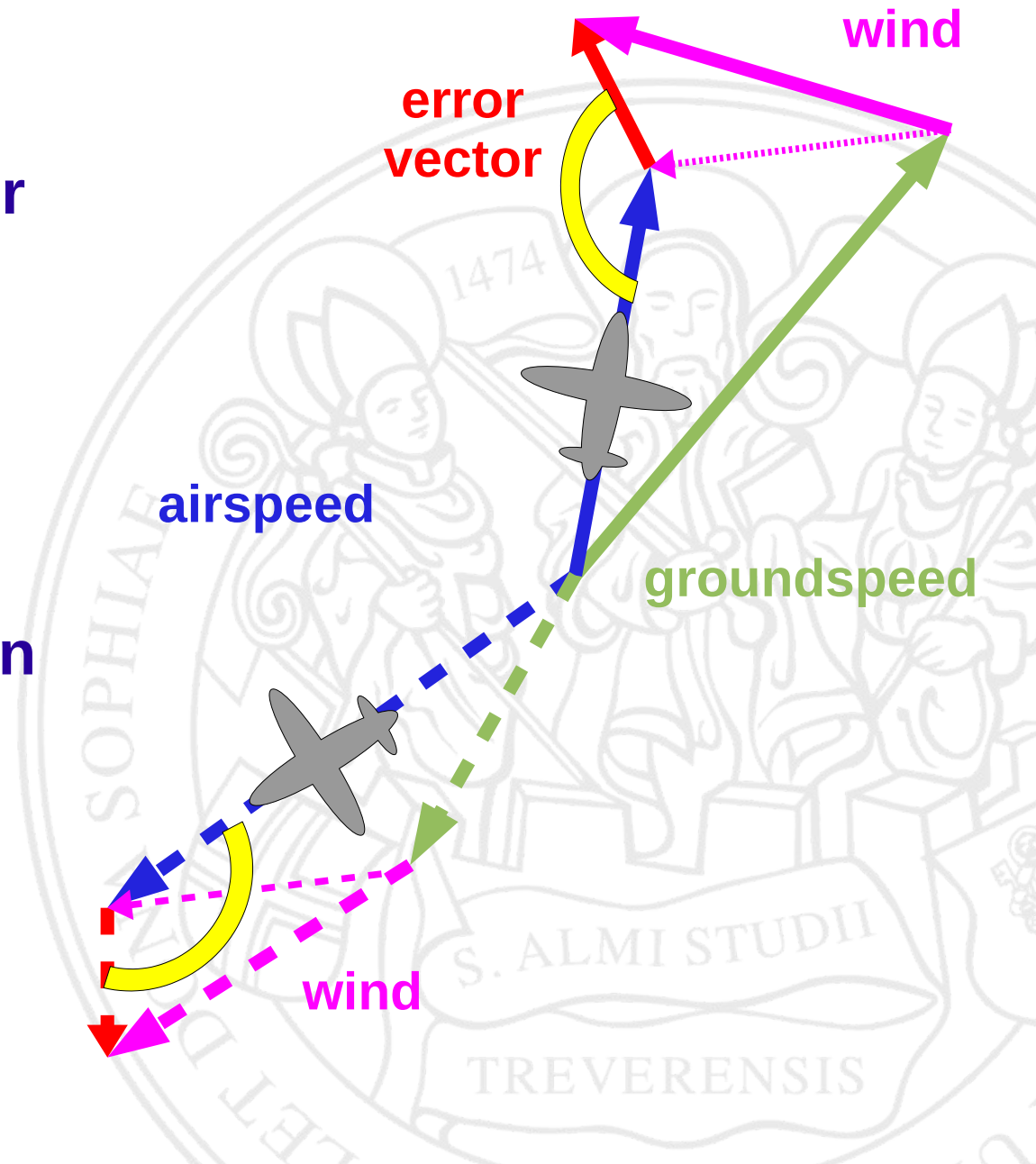
Error vector presumably
not only type dependent



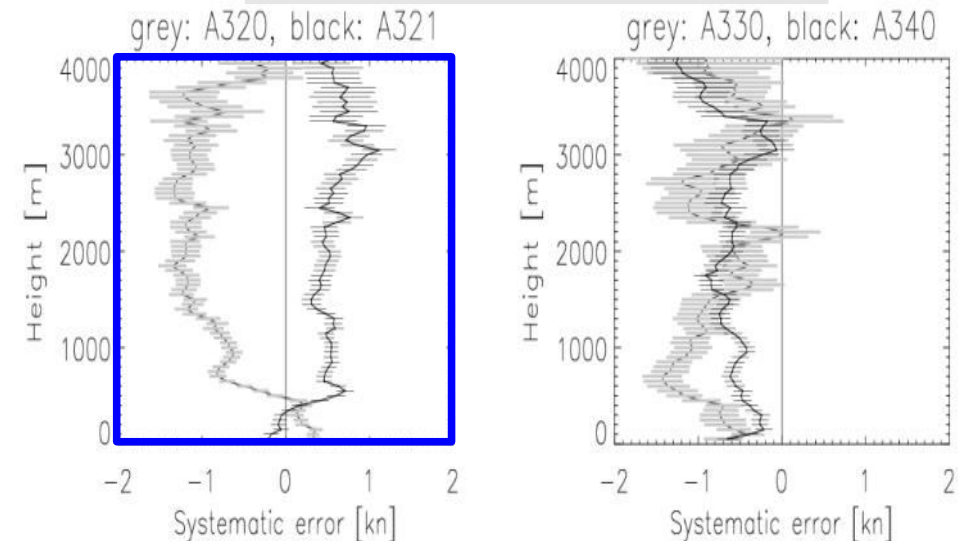
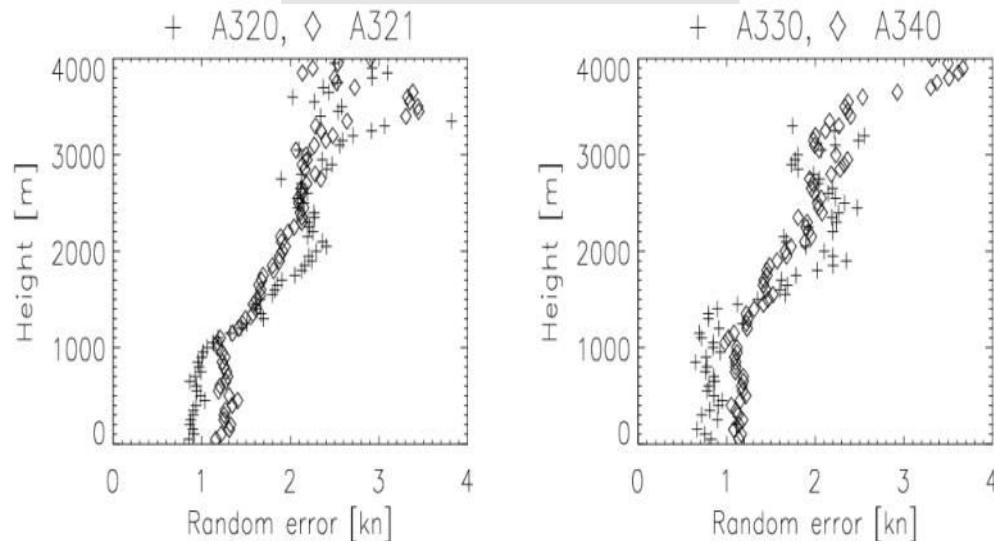
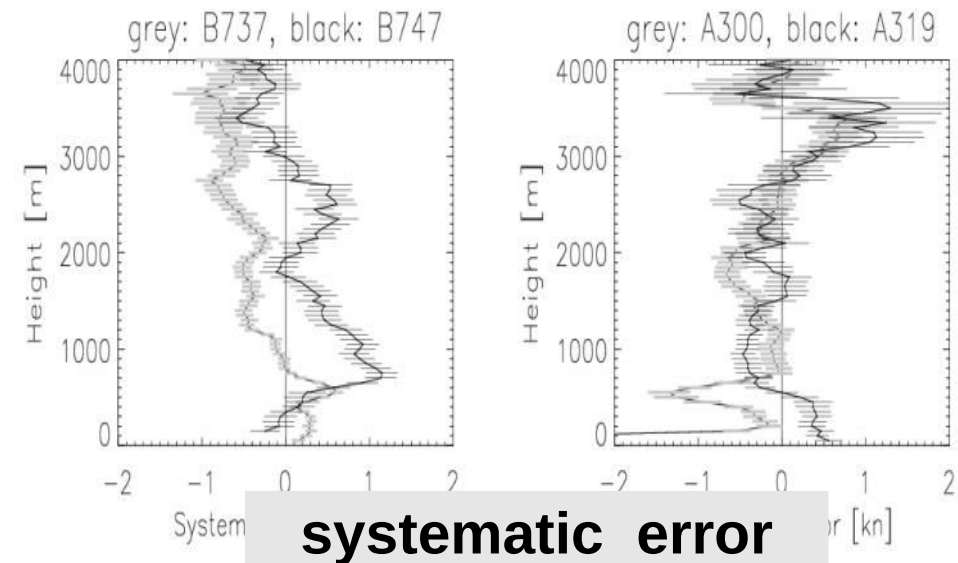
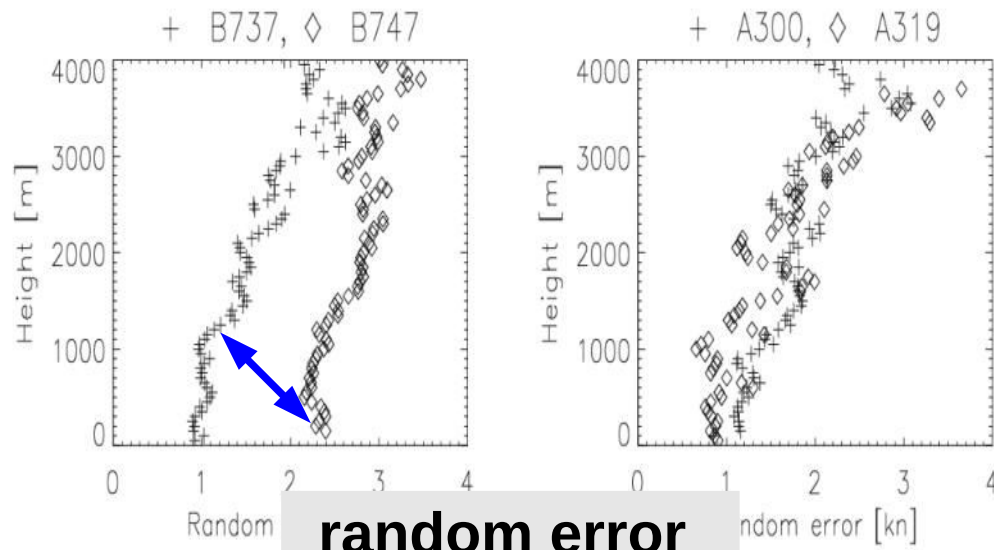


Error study: Wind error vector

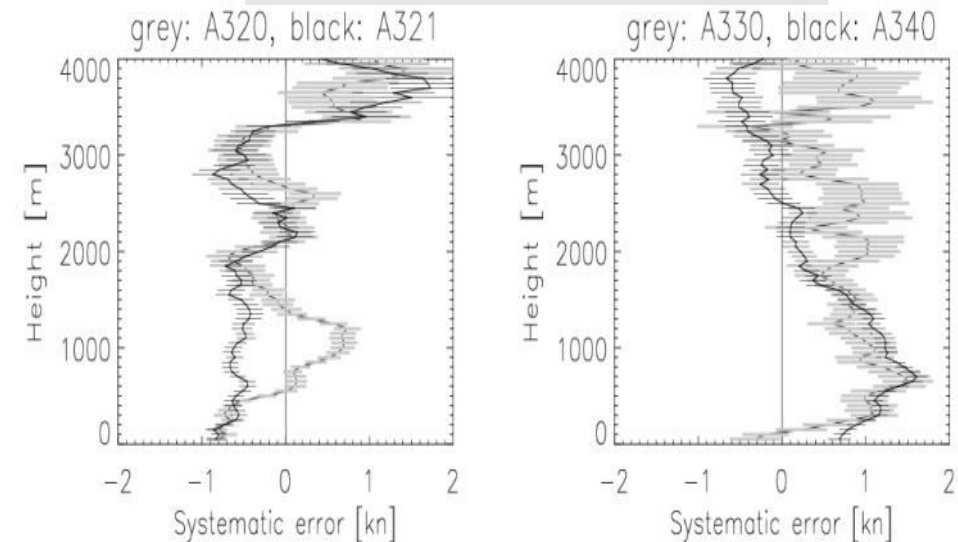
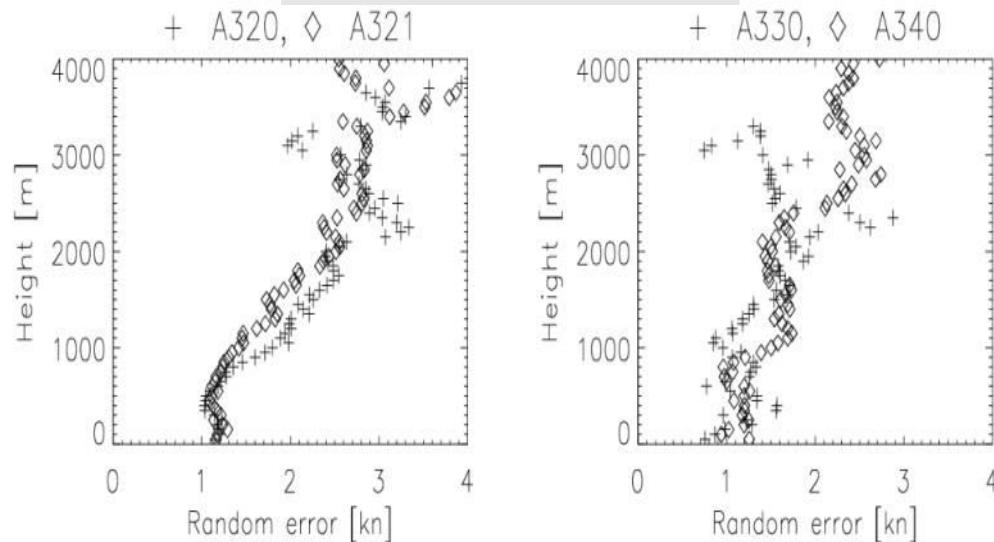
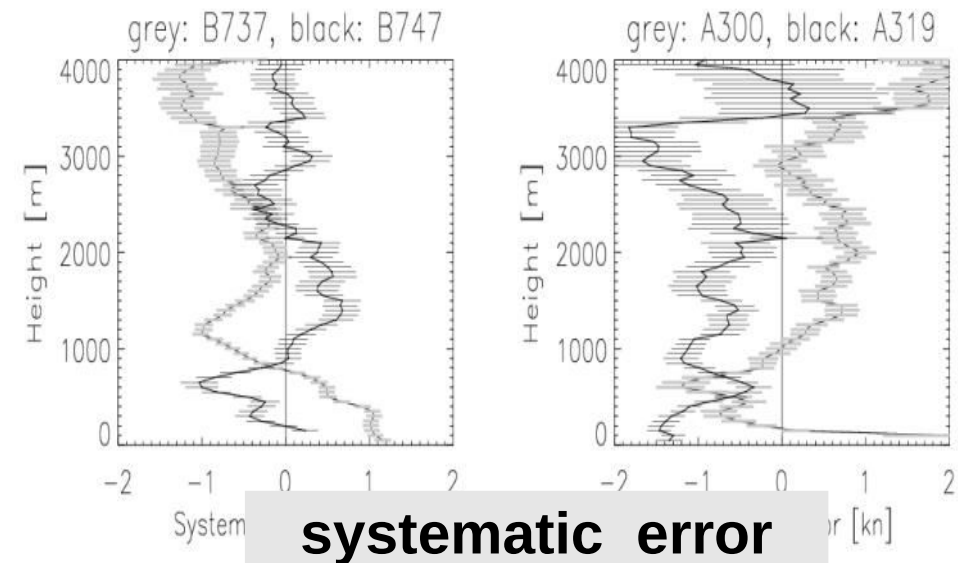
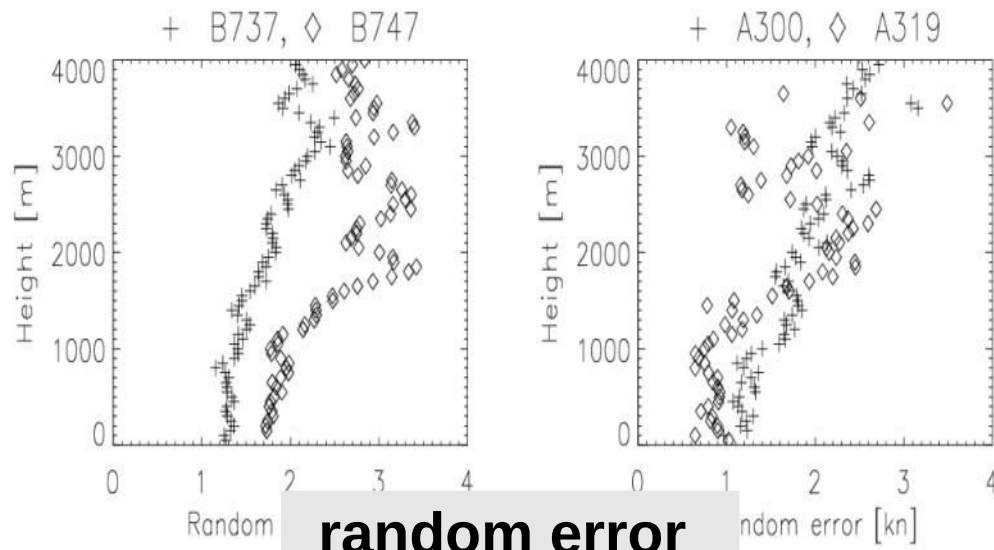
- **Wind speed error:**
Aircraft-fixed error vector
- **Errors in**
 - wind speed and
 - wind direction
- **depend on flight direction**
- **Heading**
 - needed for correction
 - not (yet) in AMDAR data

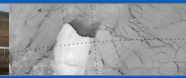


Longitudinal wind component



Lateral wind component





Influencing parameters

Roll angle

raw estimate from AMDAR

Mach number

substitute: ground speed

Heading

substitute: ground track direction ("track")

Pitch

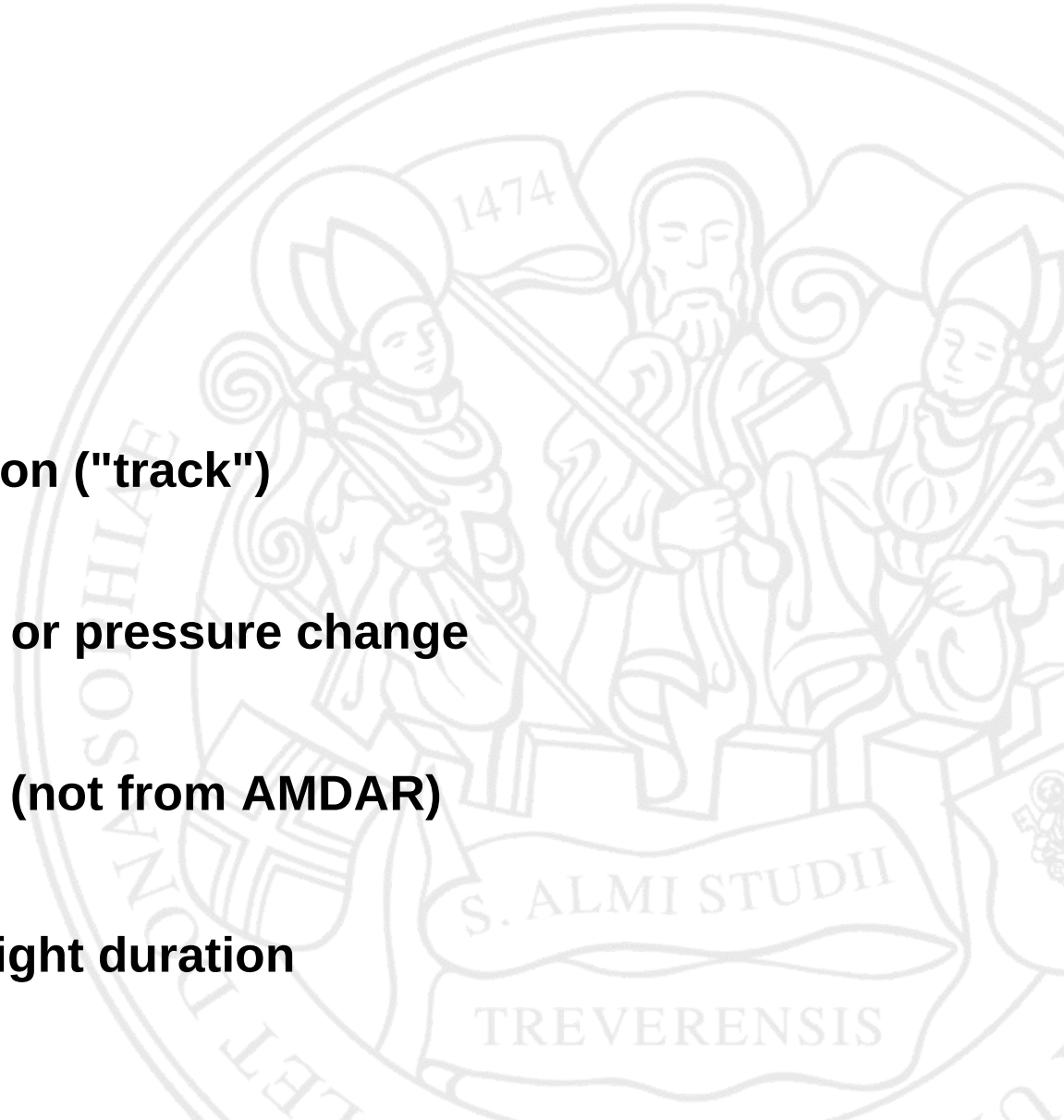
substitute: ascent/descent rate or pressure change

slip angle

substitute: lateral acceleration (not from AMDAR)

Weight-over-Delta

substitute: static pressure or flight duration





Comparison: AMDAR vs. ground-based profiler

→ Scintec AP1000 / RASS

- max. range 3000 m
- vertical profiles of
 - (virtual) temperature
 - wind speed / direction

→ data available

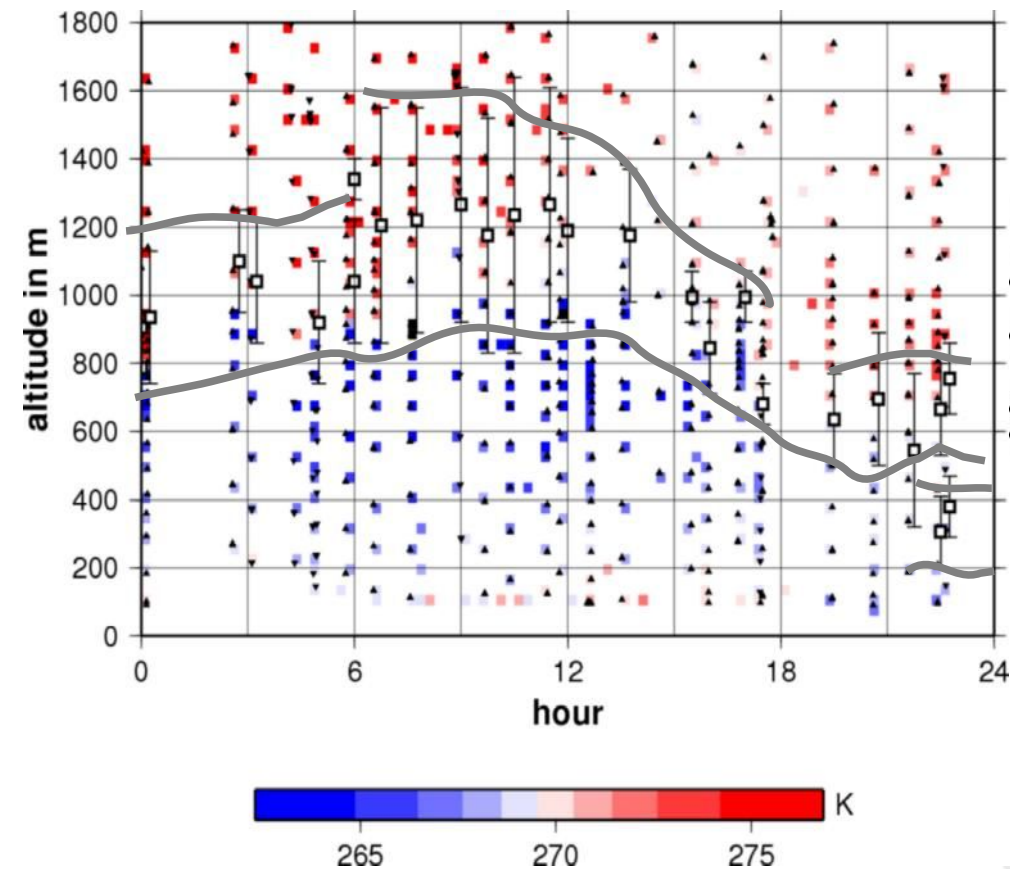
- since Dec. 2005
- 100 m to 1750 m height every 30m
- every 15 min



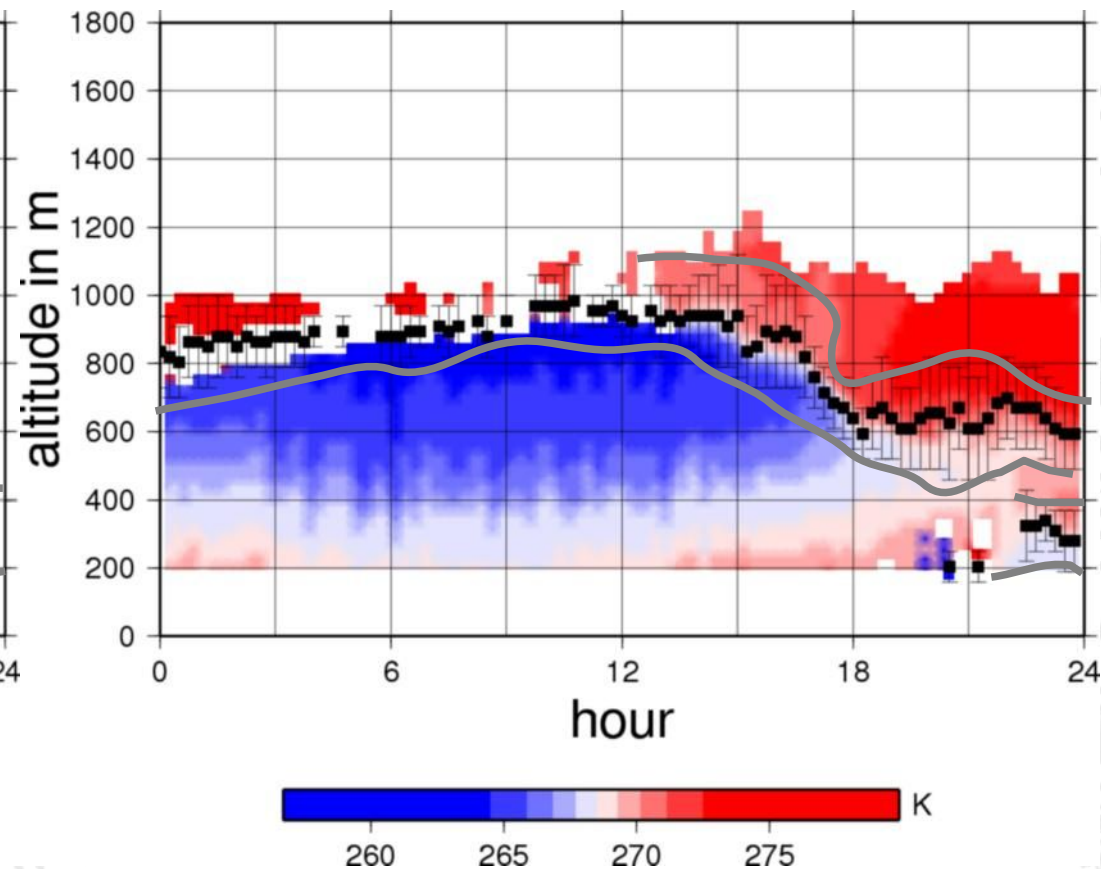
picture courtesy of Scintec AG

WT/RASS at Frankfurt

Data example: 15 Jan 2006



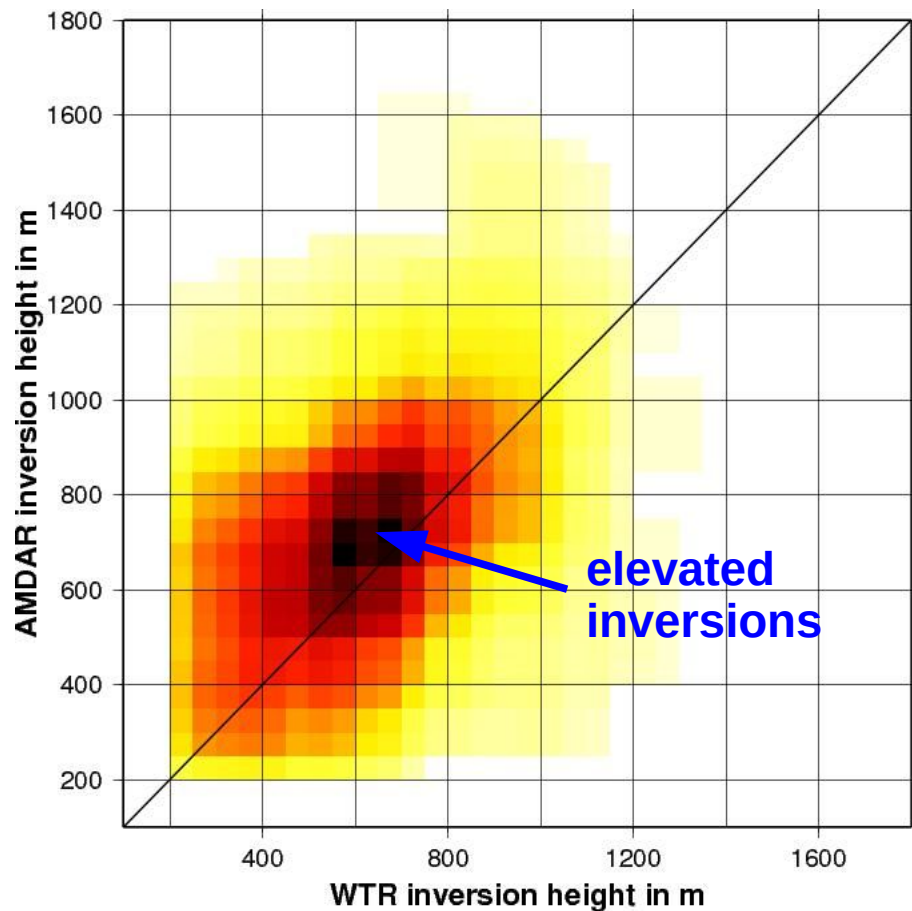
AMDAR



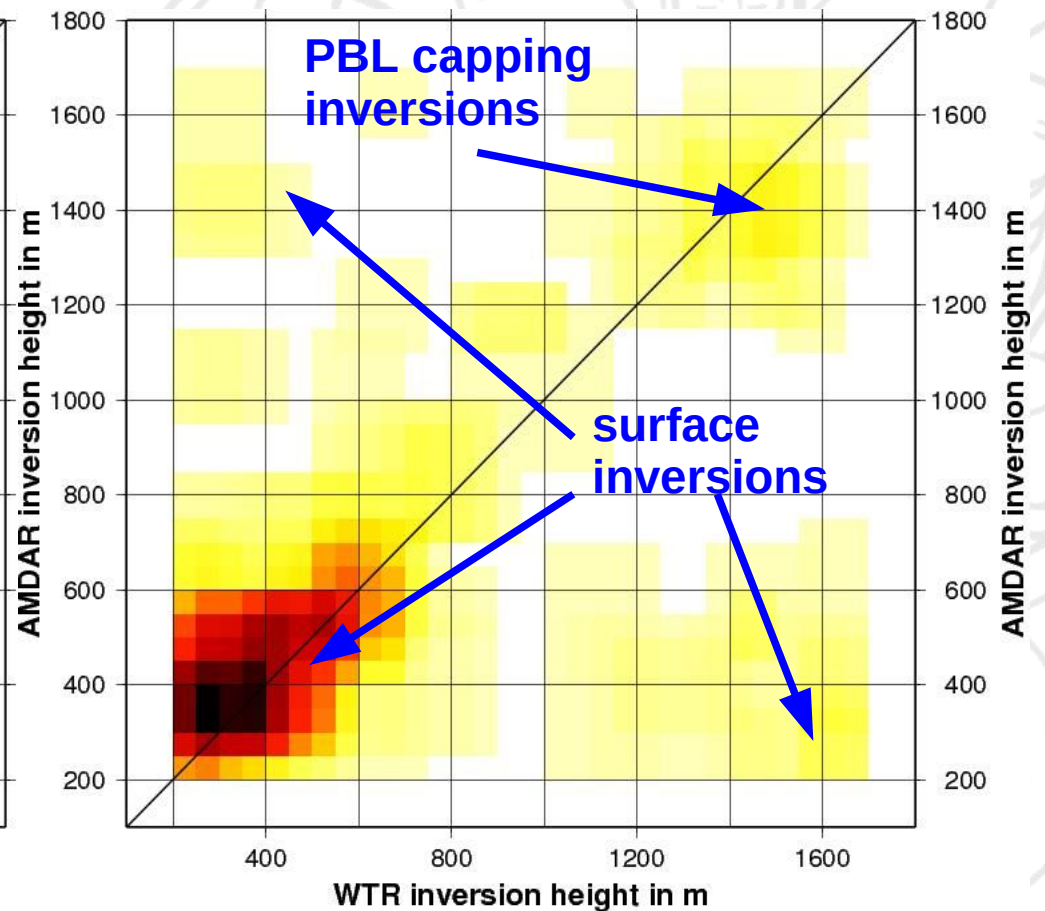
WTR/RASS

Inversion overlap

January: 1074 cases



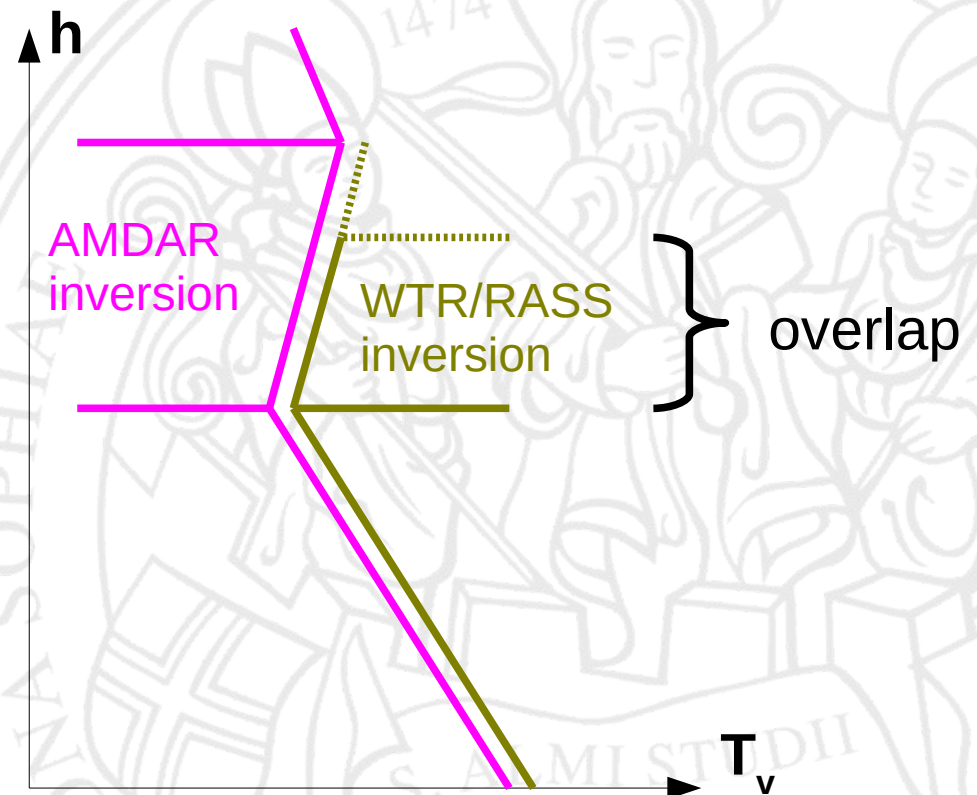
July: 111 cases





WTR/RASS range issue

- Inversion top above WTR/RASS range
 - base detected by both
 - top detected in AMDAR
 - WTR/RASS sensed inversion ends at range
 - overlap only in lower part of inversion
- shallower inversion
- downward shift of median height





Objective / Method

- **Flight dynamic data needed to model errors**
 - No flight-dynamic information transmitted
 - Radar data inaccessible
Track identification impossible (anonymous identifiers)
- **Reconstruct flight dynamics**
 - Simplified flight simulator
 - BADA performance model for horizontal / vertical speed
 - Horizontal flight path templates
(N straights, $N-1$ constant-roll turns)
 - Variational solution using cost function
 - Minimal distance to AMDAR data positions
 - Minimal flight distance and turn angle
 - Minimal difference between TAS and BADA speed schedule
 - ...

Flight simulator

→ Not straightforward

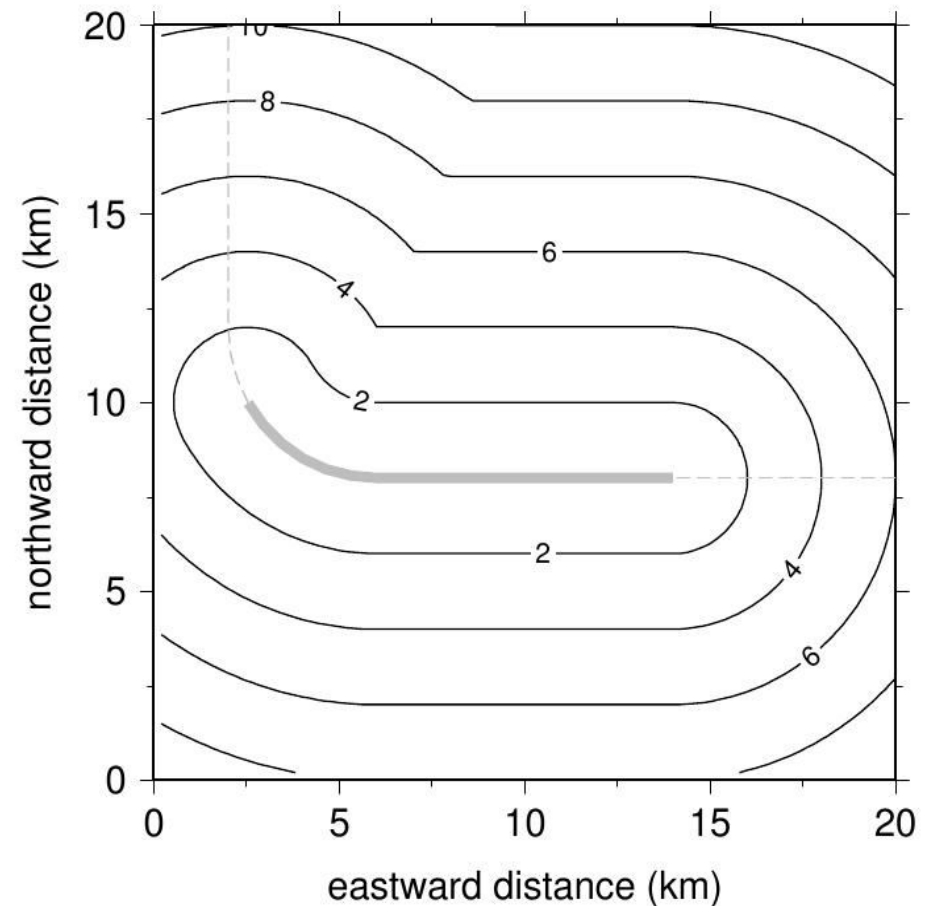
- Unsharp AMDAR reports
 - Time: seconds stripped
 - Position: seconds stripped
 - Height: small uncertainty

→ Ensure numerical stability

- Prevent Oscillations / Loops
- Allow minor heading adjustments
- Aircraft performance limits

→ Combination of

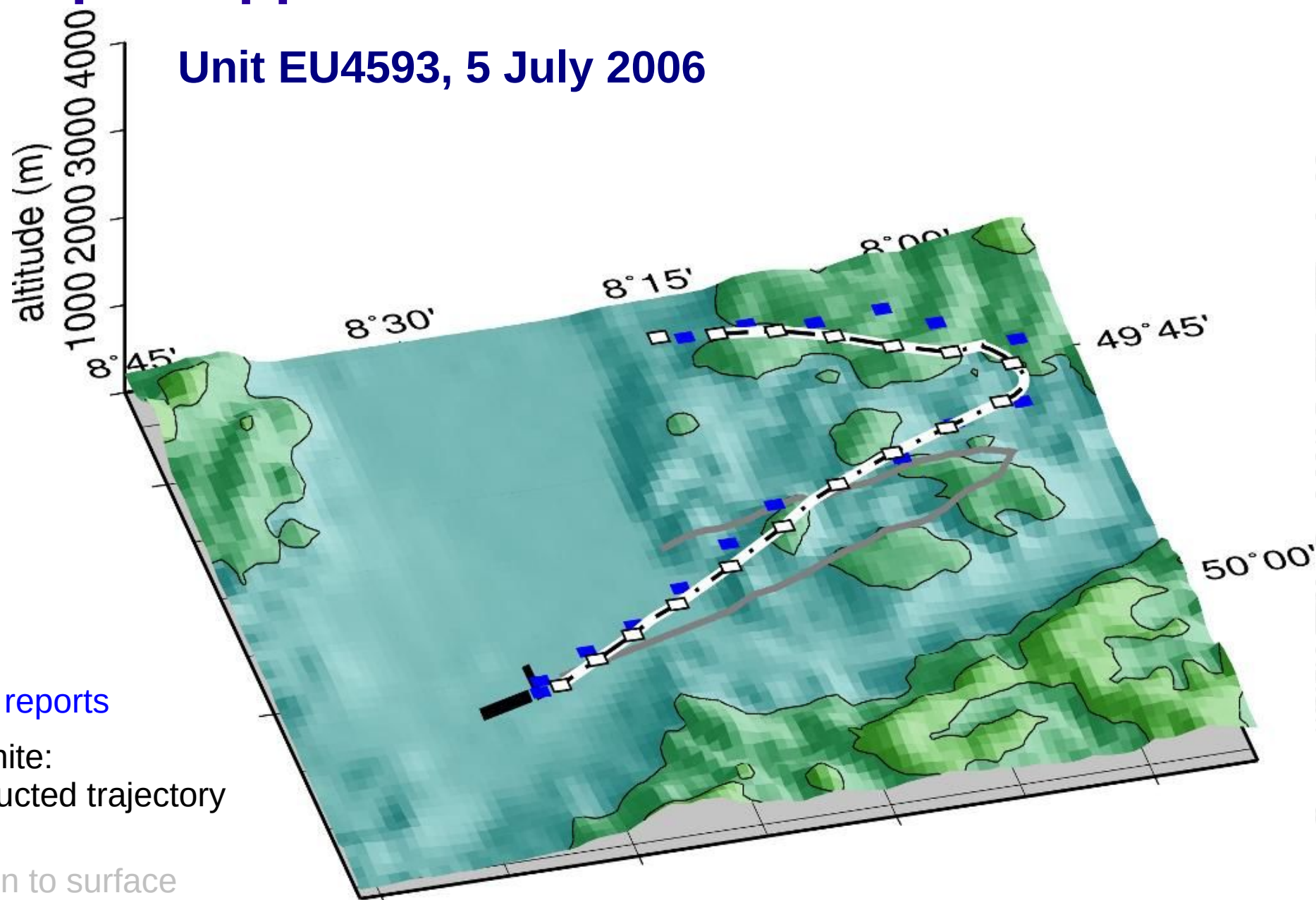
- Dimensional analyses
- Manual tuning





Example: Approach to Frankfurt/Main

Unit EU4593, 5 July 2006



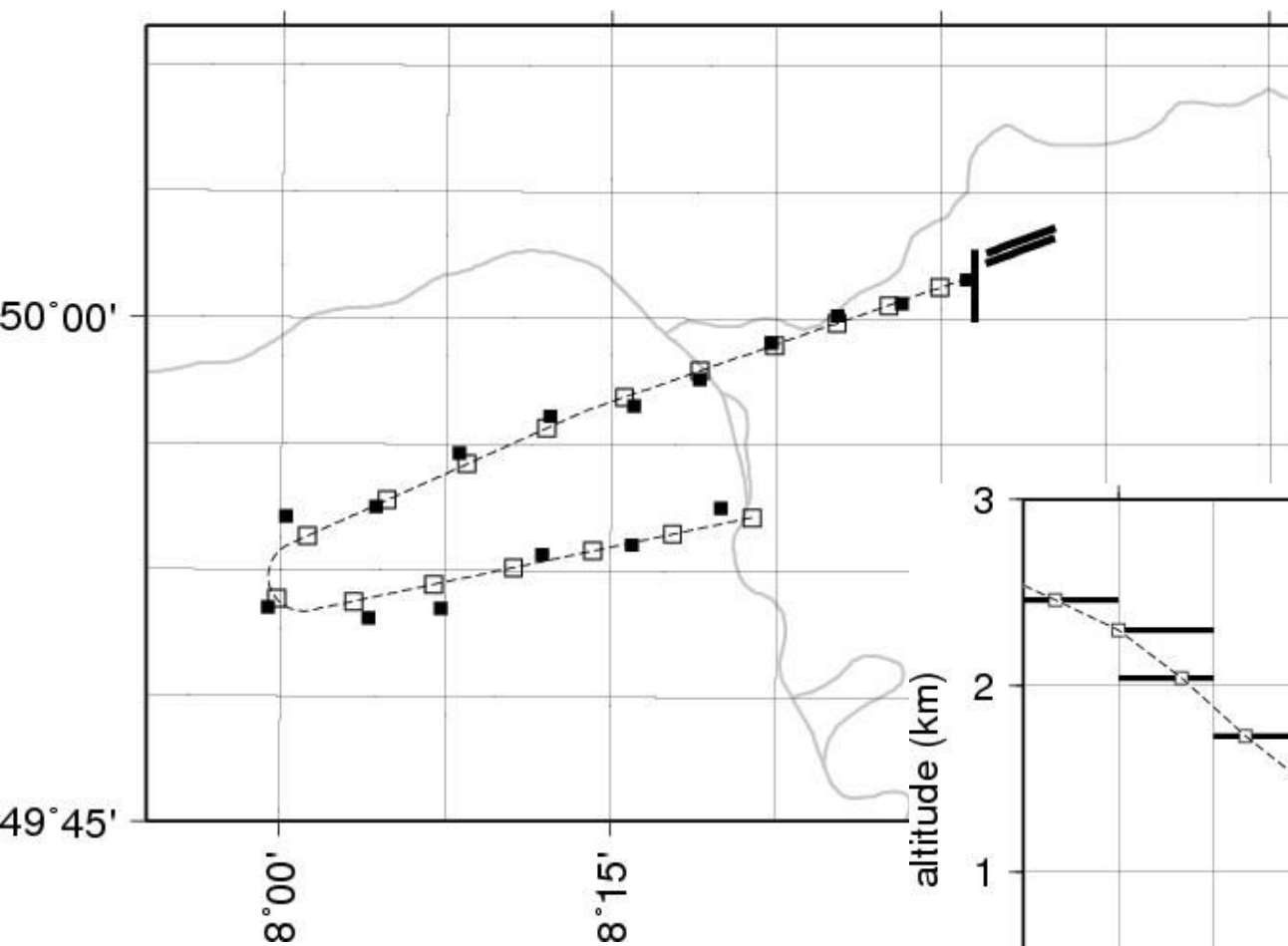
Blue:
AMDAR reports

Black/white:
reconstructed trajectory

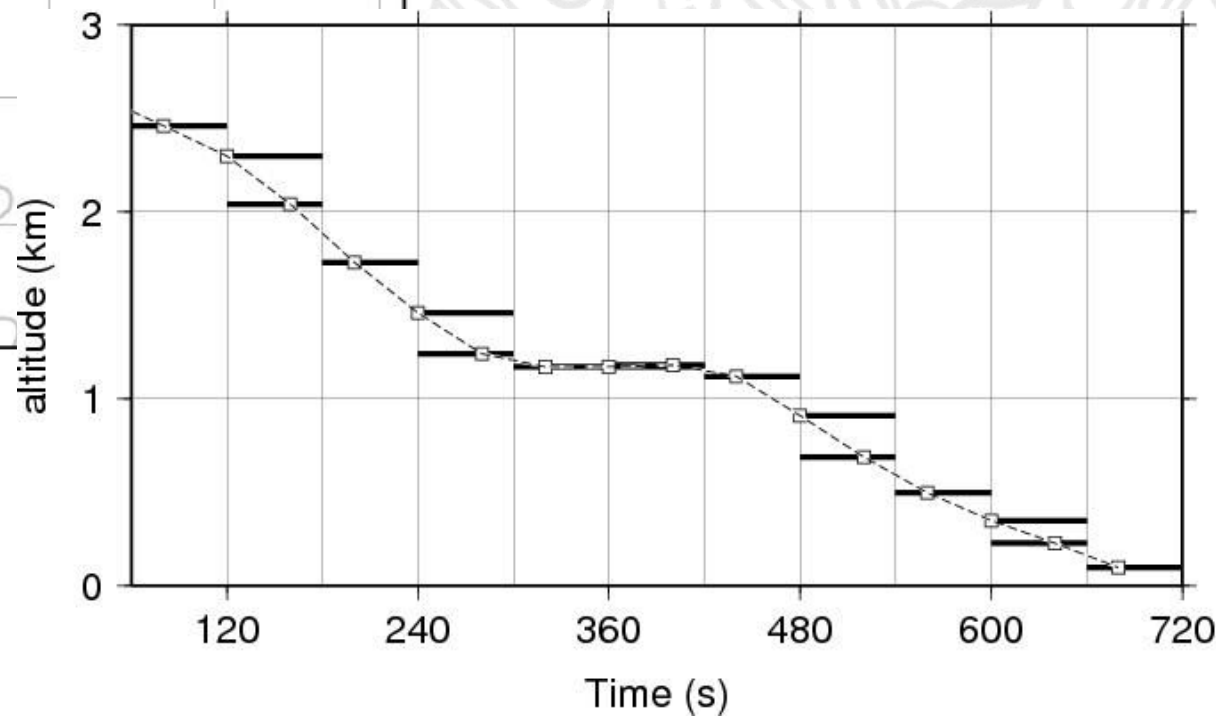
Grey:
projection to surface



Horizontal / vertical



→ Approach of
EU4593 to
Frankfurt,
5 July 2006



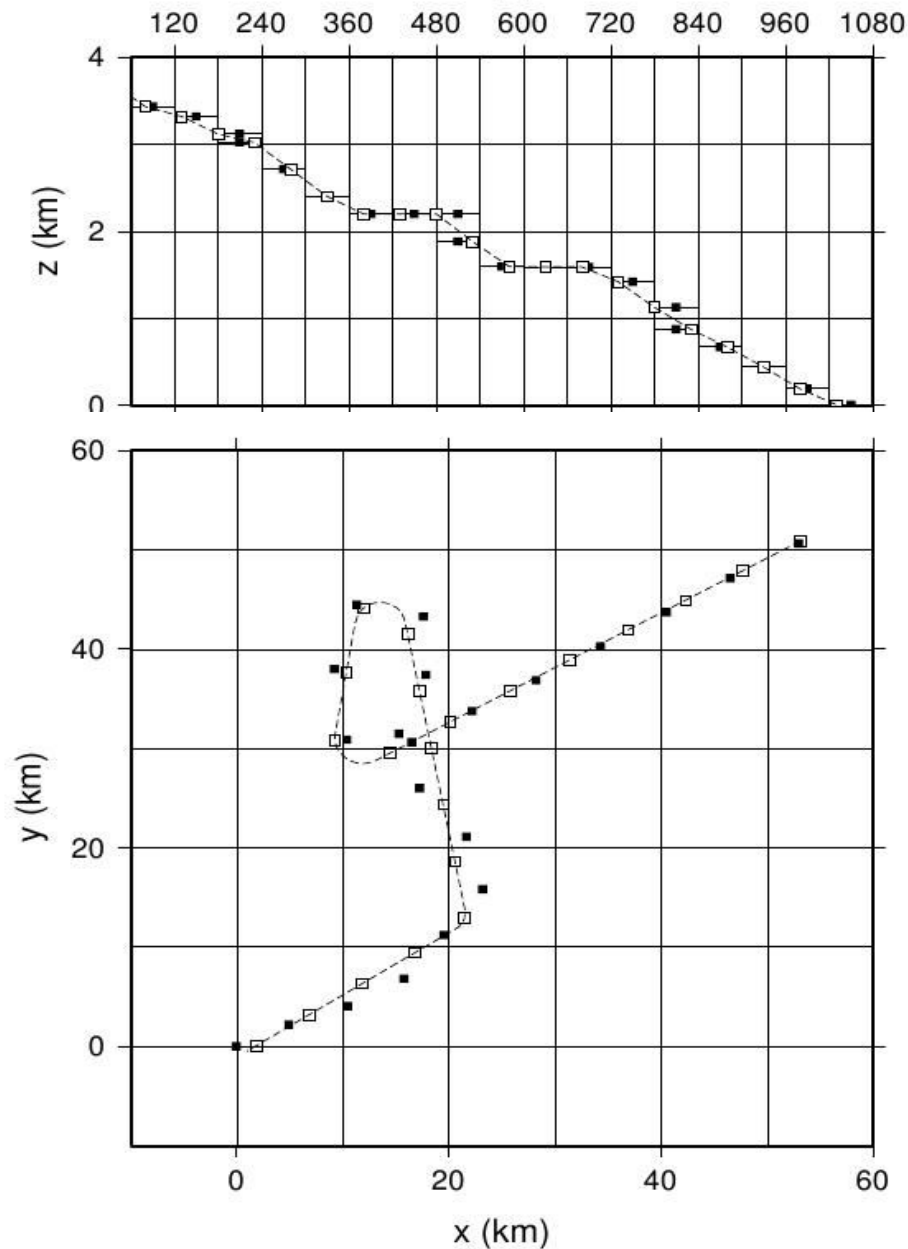
Validation

- **CARIBIC (civil aircraft for the regular investigation of the atmosphere based on an instrument container)**
- **Lufthansa A340-600 D-AIHE**
- **Records data at 1 Hz**
 - Scientific data (gas composition)
 - Meteorological data
 - Flight dynamic data
- **Pseudo-AMDAR reports generated**
 - Using DLH reporting scheme
- **Trajectory reconstructed from AMDAR**
 - Compared to actual recording

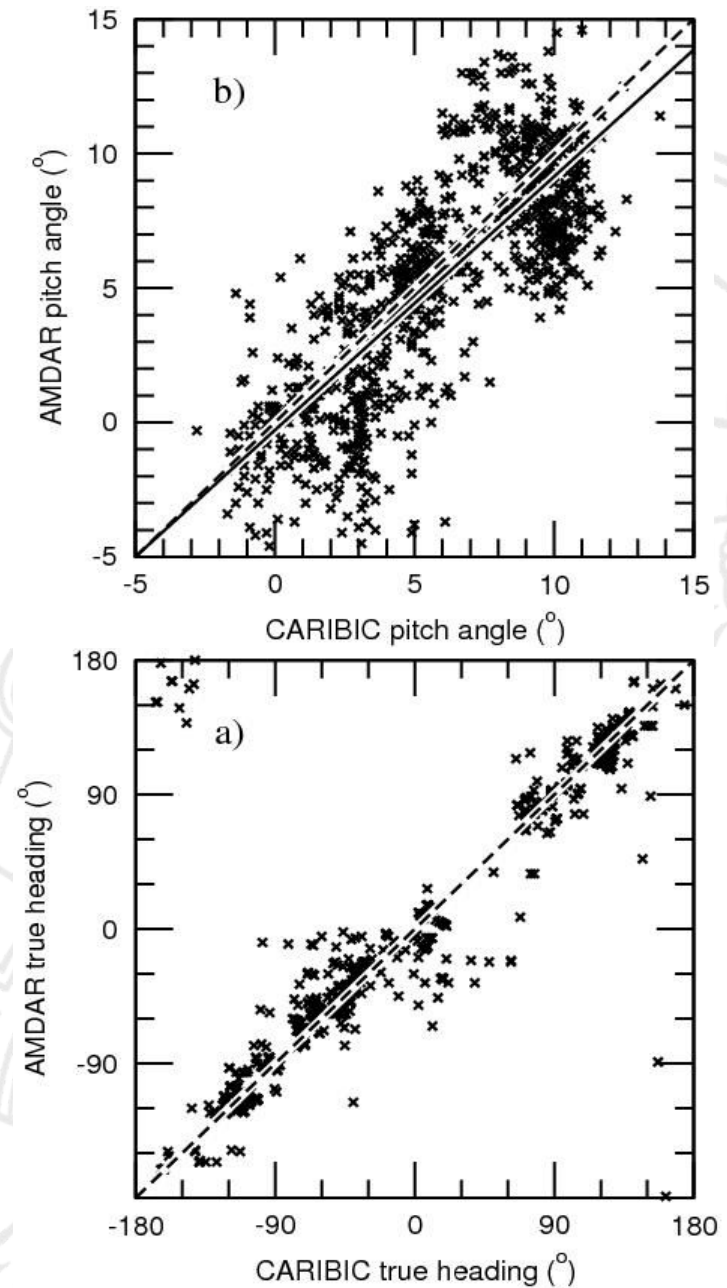


"Leverkusen" descent to Frankfurt (Main) on 2008-03-27

t (s)



Validation



Validation results

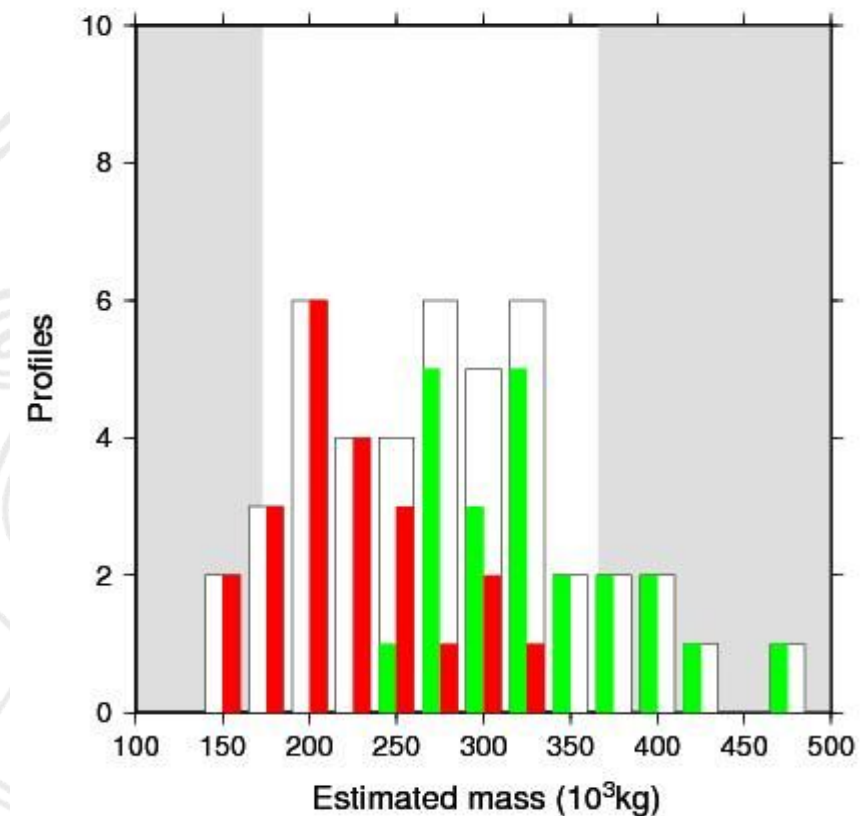
→ Very good:

→ Heading, Pitch, 3D-speed

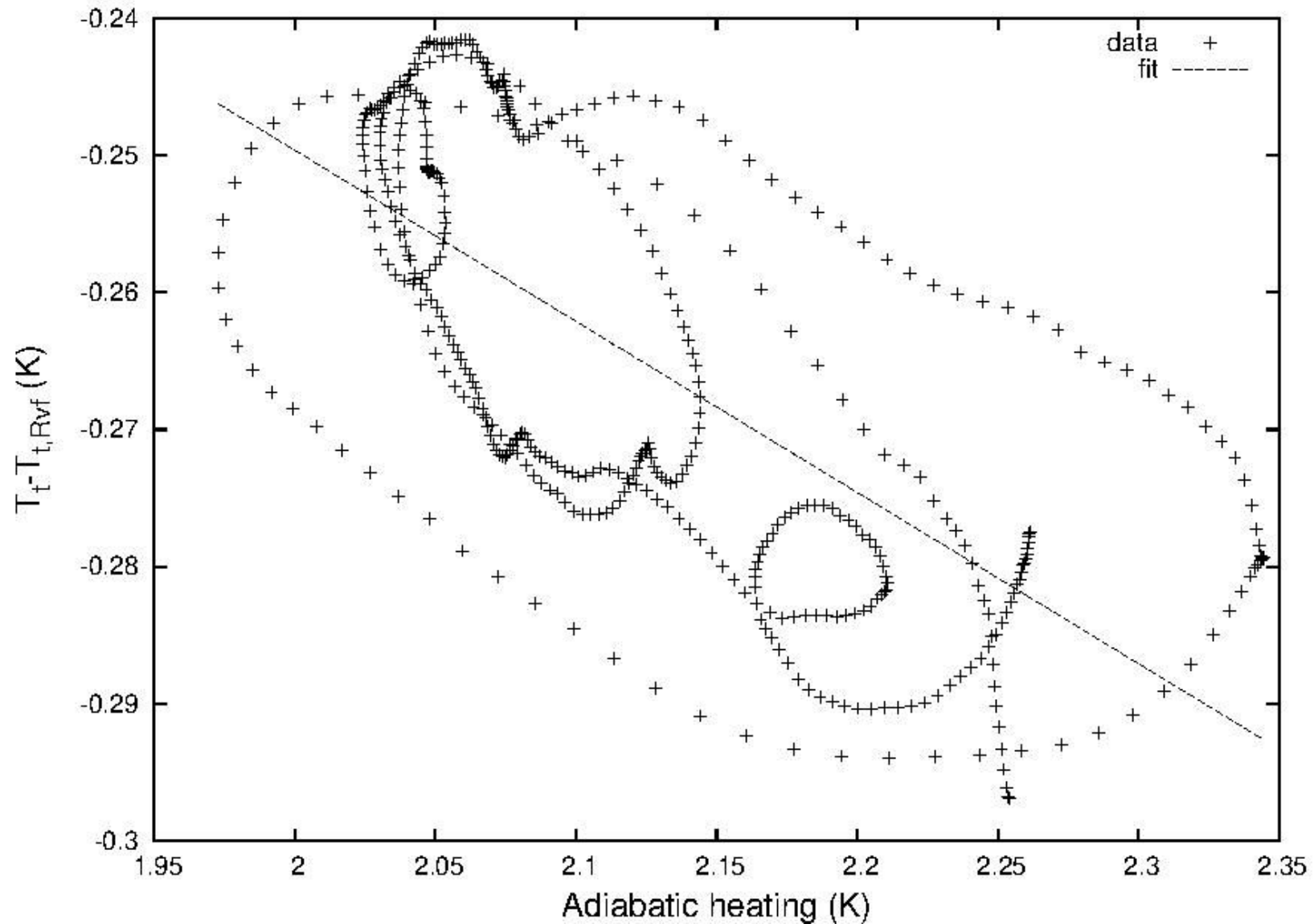
→ Fair:

→ Aircraft mass

Value	(unit)	intercept	slope	r^2
True heading	°	2.60	0.850	0.72
Pitch	°	-0.28	0.943	0.65
Roll	°	0.03	0.238	0.08
Angle of attack	°	1.01	0.655	0.15
w	ms^{-1}	0.35	0.965	0.94
v_{as}	ms^{-1}	1.28	0.943	0.68
v_{gs}	ms^{-1}	18.1	0.881	0.79



Example: Regression p,TAS vs T-offset



Conclusions

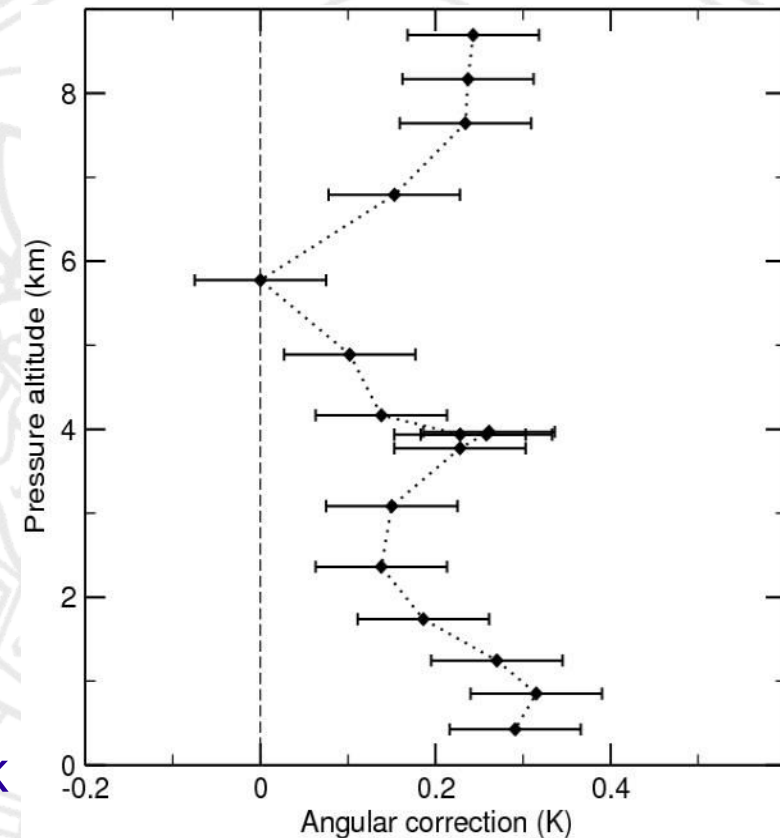
→ Trajectory reconstruction

- Physically consistent
- Flight dynamic retrievable
- Moderate computational costs

→ Validation

- DLH A340-600 and DLR Learjet
- Very good for most variables
 - Large scatter for roll and angle of attack
- Mean agreement reasonable

→ Valid candidate for error modeling



Outlook

→ Statistical model

- errors as function of flight state
- major component analysis

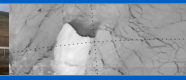
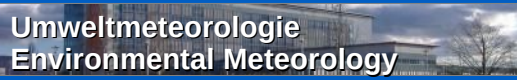
→ Feasibility of ADS-B

- for wind measurement without Mode-S radar position data
- for developing countries

→ Need for graduants

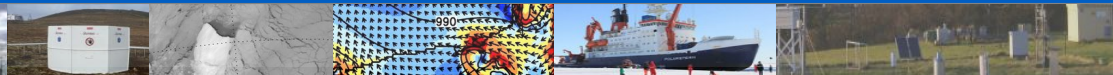
- increasingly (!) deterred by required programming





Thank you! Questions?





References

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- Haering, Edward A. Jr. (1995): "Airdata Measurement and Calibration", NASA Technical Memorandum 104316, National Aeronautics and Space Administration (NASA) Dryden Flight Research Center, Edwards, CA, USA.
- Palacios, F. et al. (2014): "Stanford University Unstructured (SU2): Analysis and Design Technology for Turbulent Flows." 52nd AIAA Aerospace Sciences Meeting - AIAA Science and Technology Forum and Exposition, SciTech 2014, doi: 10.2514/6.2014-0243.