



CIMR COPERNICUS IMAGING MICROWAVE RADIOMETER

Strategies to mitigate the impact of RFI for the Copernicus Imaging Microwave Radiometer (CIMR)

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RFI2022, ECMWF 14-18th February, 2022

ESA-DEVELOPED EARTH OBSERVATION MISSIONS

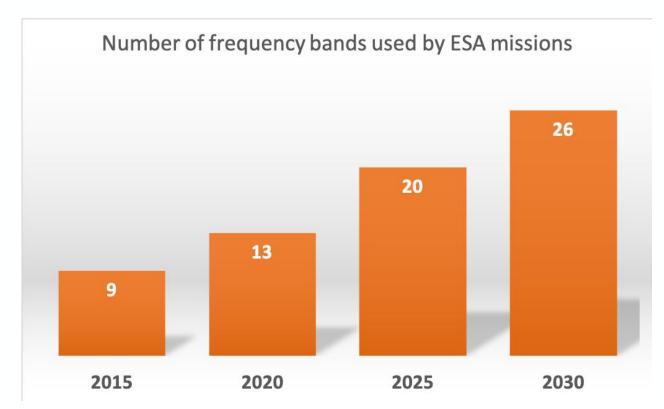




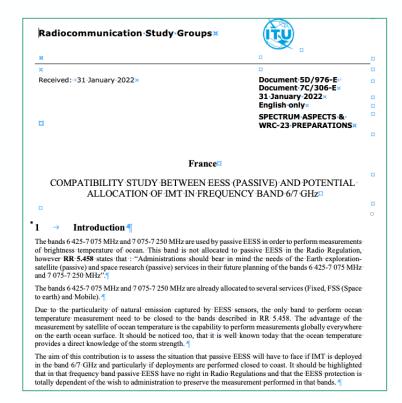


Increasing use of the spectrum





This implies an improved ability to observe the Earth system.



Because EO sensors operate outside allocations, it is difficult to argue for their protection in frequency management fora, and they cannot claim protection from the RFI they experience.

But it also implies more RFI issues and more involvement in frequency regulatory matters.



CIMR: RFI issues



- RFI is present in all CIMR frequency bands (as for most microwave radiometer instruments)
- RFI issues are expected to increase, as the use of the spectrum increases
- In the design phase, it is important to take RFI into account to:
 - Ensure survivability of the instrument;
 - Ensure high rejection levels outside the allocated band;
 - Develop RFI detection strategy;
 - Develop RFI mitigation approaches;
 - Develop RFI monitoring approach (evidence for ITU and spectrum pollution management)
 - In Phase E2 operations, it is important to implement ground based RFI mitigation and monitoring techniques based on the CIMR data → esach sensor 'sees' RFI in its own way
- RFI in some frequency bands can be reported to the ITU (e.g. in the bands where all emissions are prohibited by article 5.340 of the Radio Regulations). This has decreased RFI in L-band!
- Possible to coordinate with other space agencies to report RFI: now RFI in L-band are reported by SMOS and SMAP around the same time.

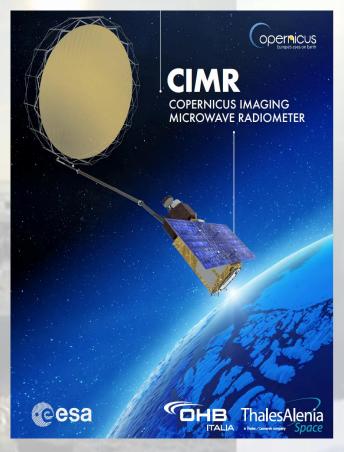


The Copernicus Imaging Microwave Radiometer (CIMR)





The European Commission and the High Representative of the Union for Foreign Affairs and Security Policy issued to the European Parliament and the Council, on 27 April 2016, a joint communication that proposed "An integrated European Union policy for the Arctic"



Polar Oceans are fundamental to understanding the global environment

CIMR is designed to:

- Prevent the anticipated Gap in capability
- Be "ready" for an ice free Arctic
- Key variables: Sea Ice Concentration, Sea Surface Temperature, thin Sea Ice Thickness, Sea Surface Salinity, Wind Speed, soil moisture...
- Low frequency/High Spatial resolution (5–15 km)
- Measurements every ~6 hours in the Polar regions, no hole at the pole
- 95% global coverage every day for application in all Copernicus Services
- Directly addresses the EU Arctic Policy.
- A 'Game Changer' for Copernicus

























































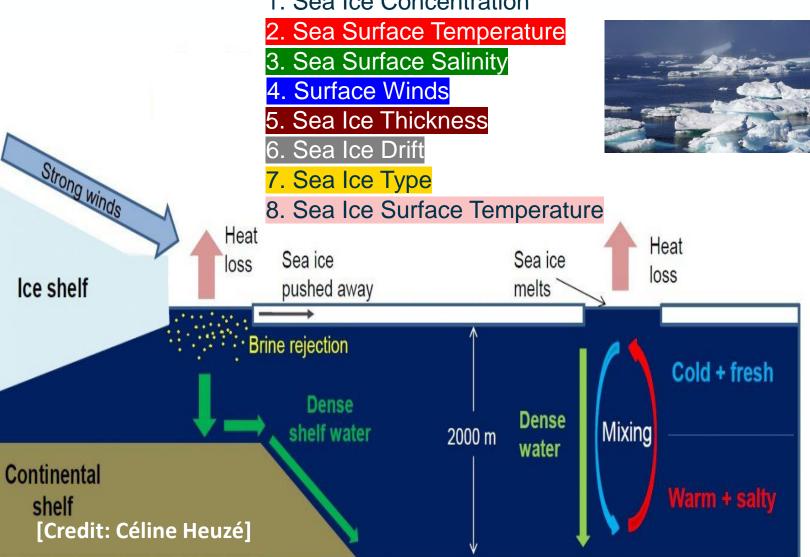
Ice shelf

shelf

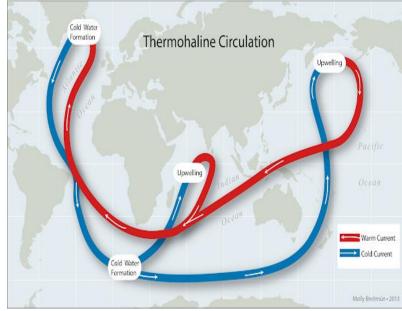
CIMR: Cryosphere-ocean-atmosphere processes

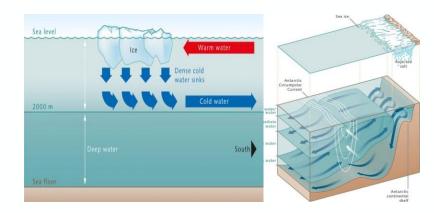


1. Sea Ice Concentration







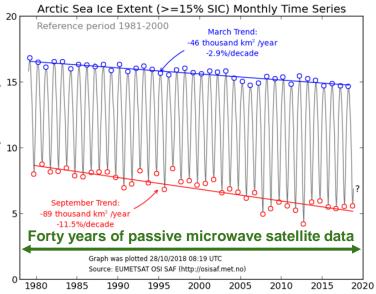




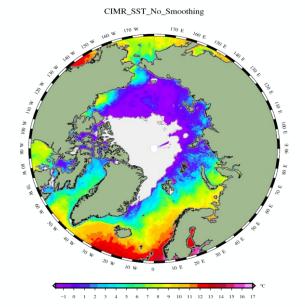
Level-2 Measurement Products



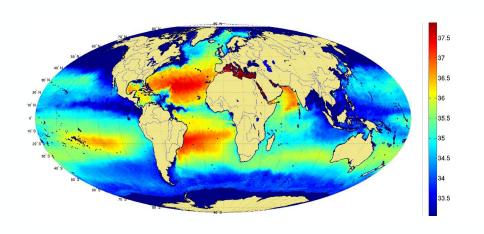
Sea Ice Concentration



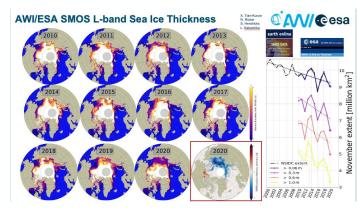
Sea Surface Temperature



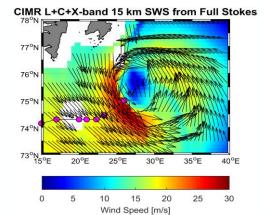
Sea Surface Salinity



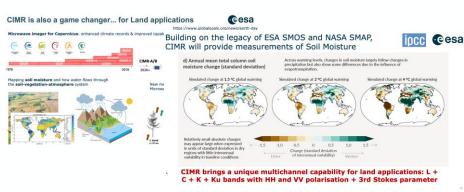
Thin Sea Ice thickness



Surface Wind over ocean



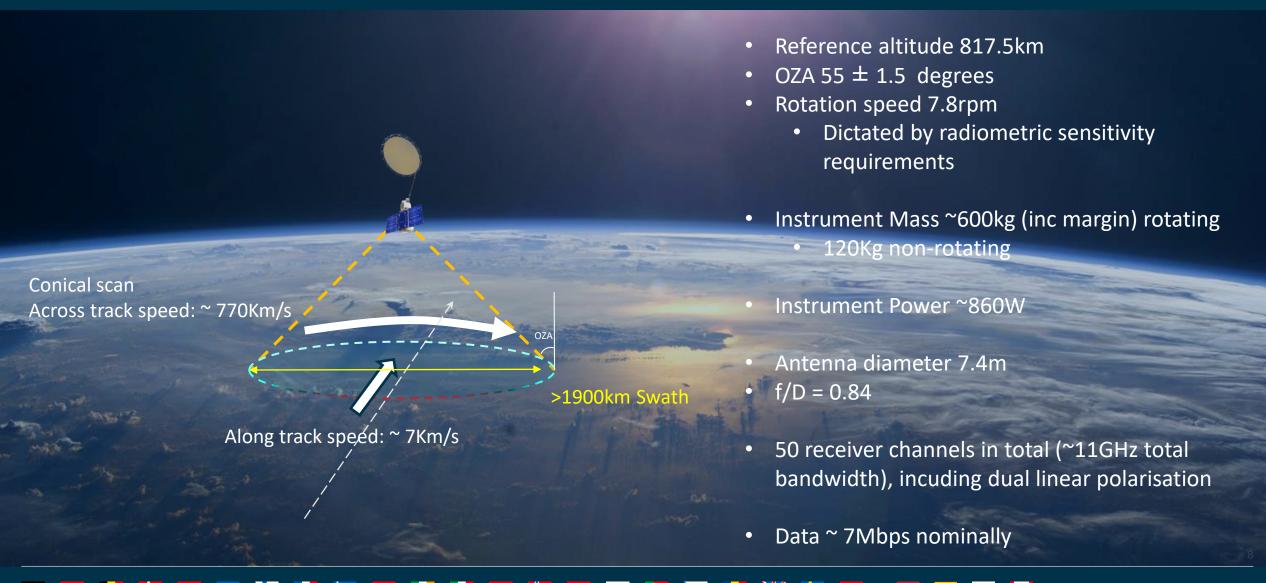
Sea Ice Drft, ice type, snow, Vegetation, soil moisture...





The CIMR Payload Overview







opernicus CIMR Orbit design



CIMR is to be placed in a 06:00 sun synchronous dawn-dusk orbit

CIMR flies 'ahead' of MetOp-SG(1B)

- +/- 10 mins difference in Arctic
- Focus on the Arctic region
- No "hole at the pole"
- Minimise daily eclipse periods and mitigate the impact of thermoelastic distortion,
- Maximise power generation,
- Minimise the complexity and size of the solar array.
- Maximise the colocation between CIMR measurements and MetOp-SG(B) within ±10 minutes in the polar regions



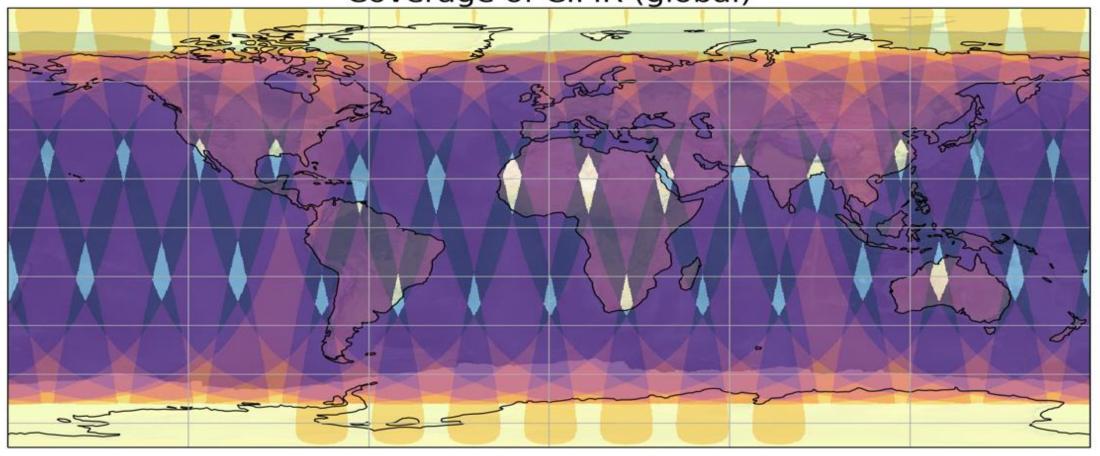


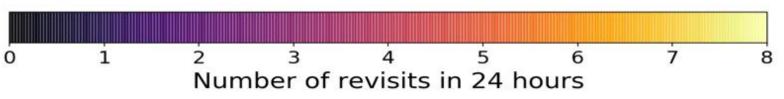
Coverage and Revisit

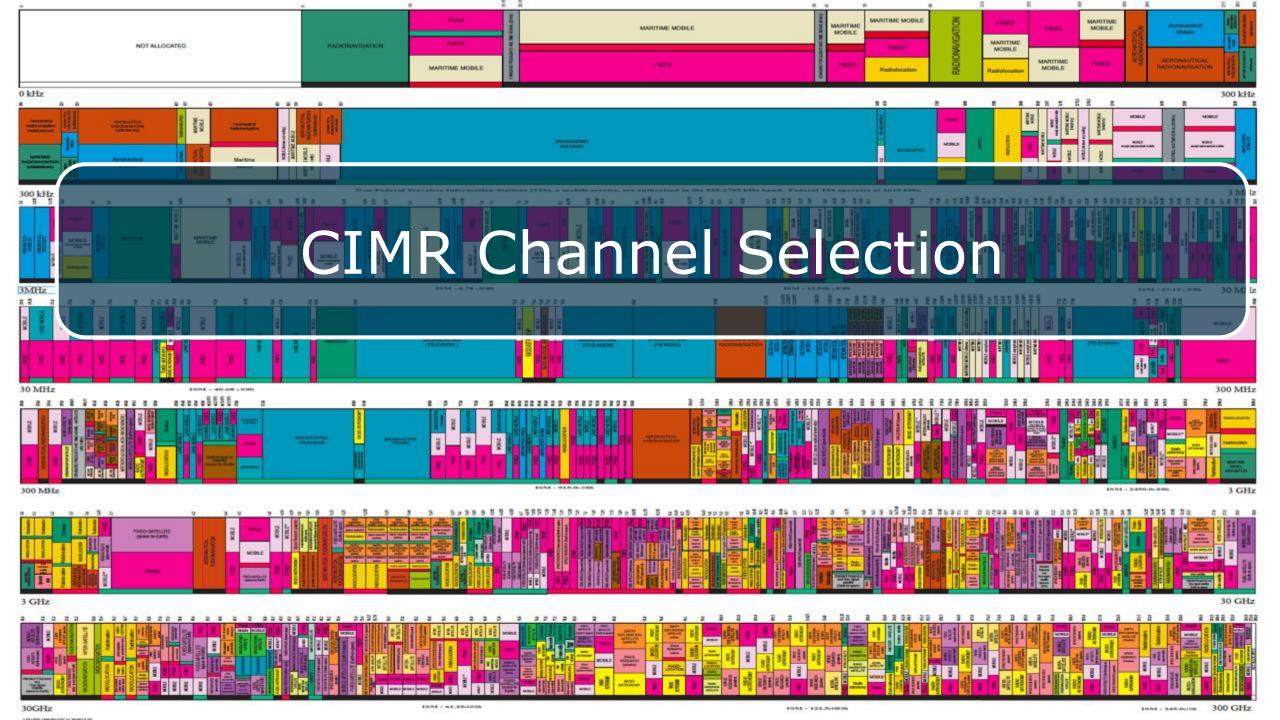


(~95%/day Global coverage, 1 satellite; 99% coverage after 1.5 days)

Coverage of CIMR (global)









opernicus Channel selection



1.4145 GHz: SIT, SIC, SSS, WS, SM, SD

6.9 GHz: SIC, SST, SIT, IST, WS, SID,

SM, SD

10.65 GHz: SST, PCP, WS, SD, SM

18.7 GHz: TCWV, LWP, PCP, SIC, SD, SM,

SID

36.5 GHz: SIC, SST, LWP, TCWV, PCP,

SIC, SWE, SD

SIC = Sea Ice Concentration, SST = Sea Surface Temperature, SIT = Sea Ice thickness. SSS= Sea Surface Salinity,

WS = Wind speed,

LWP = Liquid Water Path,

TCWV = Total Column-liquid Water Vapour,

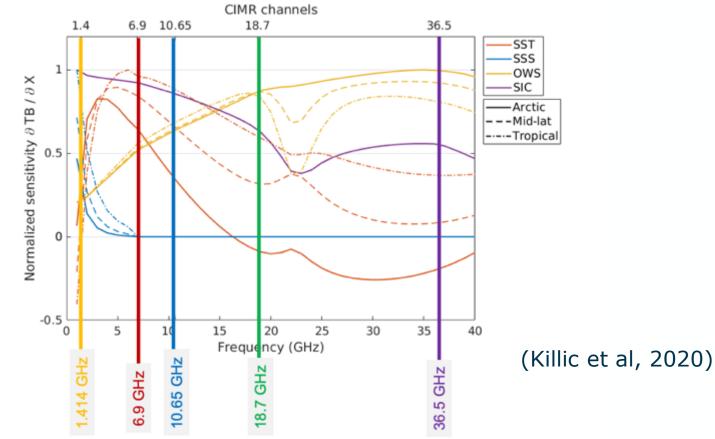
SD = Snow Depth,

SM = Soil Moisture,

SWE = Snow Water Equivalent,

SID = Sea Ice Drift,

PCP=precipitation

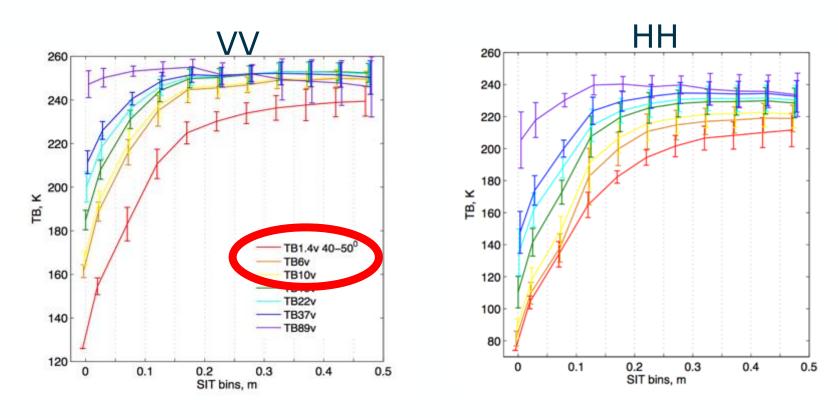


Channels (GHz, Full Stokes):	1.4	6.9	10.65	18.7	36.5
Resolution (km):	<60	≤15	≤15	≤5.5	≤5 (g:4km)
NEΔT (K @150K):	≤0.3	≤0.2	≤0.3	≤0.4	≤0.7
Tot. Standard Uncertainty(K):	≤0.5	≤0.5	≤0.5	≤0.6	≤0.8



Sea Ice Thickness channel selection





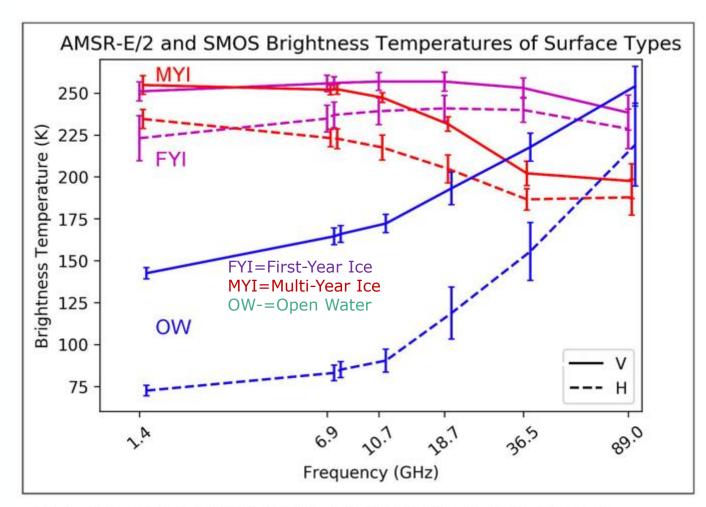
Polarized brightness temperatures as function of sea ice thickness for various frequencies (from *Heygster et al,* 2014).

The best performing frequency for thin sea ice thickness determination is 1.4 GHz.



Sea Ice Concentration: the role of L-, C/X, K/Ka-bands





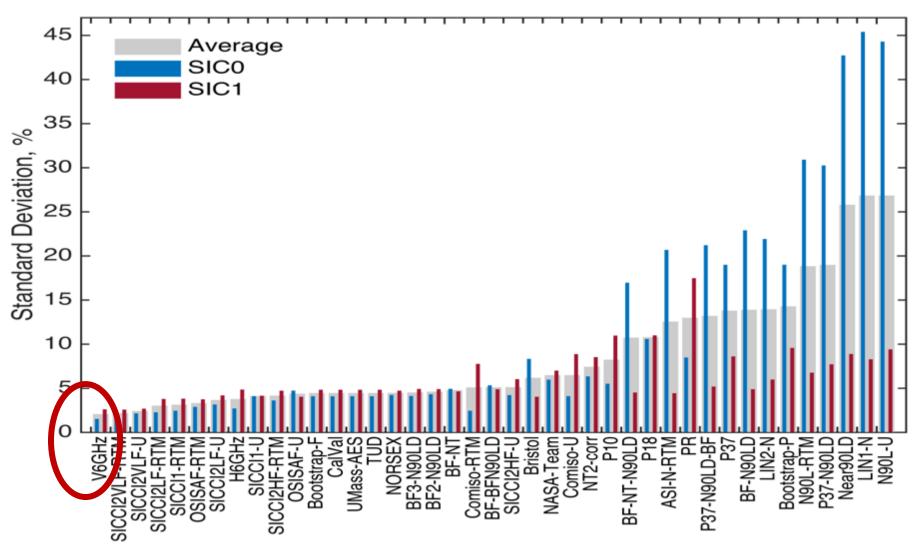
Lu, J. and Heygster, G.: AMSR-E/2 and SMOS Brightness Temperatures of Surface Types, , doi:10.6084/m9.figshare.7370261.v2, 2018.

- On a single-channel basis, we want to use low-frequencies because:
 - high dynamic range between Open Water and sea-ice.
 - limited dynamic range between Multiyear Ice and First-Year Ice.
- The best SIC algorithms involve Ka-band: high spatial resolution<5 km achievable



Sea Ice Concentration (SIC) algorithms





The best algorithm uses C-band

But...C-band has a large footprint...

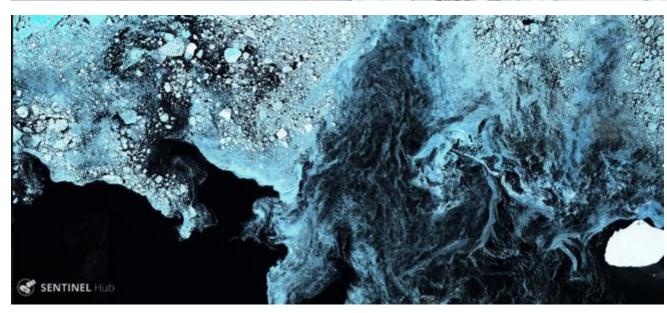
opernicus Sea Ice spatial characteristics are complex.









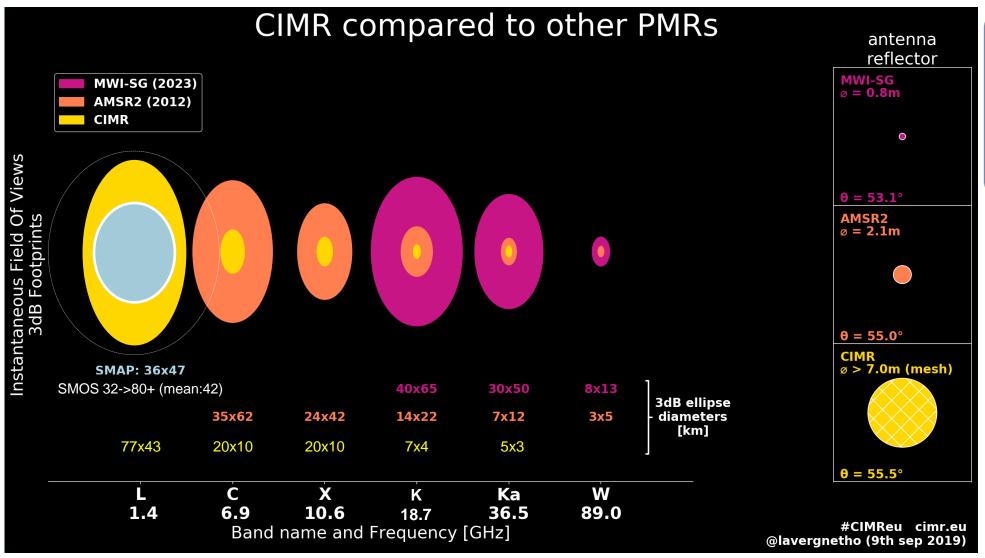


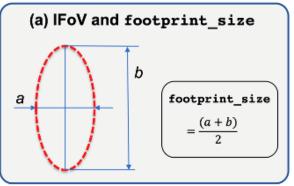




opernicus CIMR -3dB projected IFoV and footprint size







footprint size:

L: <60 km

C: ≤15 km

X: ≤15 km

K: ≤ 5.5 km

Ka: ≤5 (g:4) km



opernicus Engineering model of the 8m diameter antenna reflector









↑Antenna boom during deployment testing. © HPS GmbH

← First automatic motorised deployment test of the European LDR. © LSS GmbH

https://phi.esa.int/automatic-unfurling-of-european-large-deployable-reflector-successfully-demonstrated/











































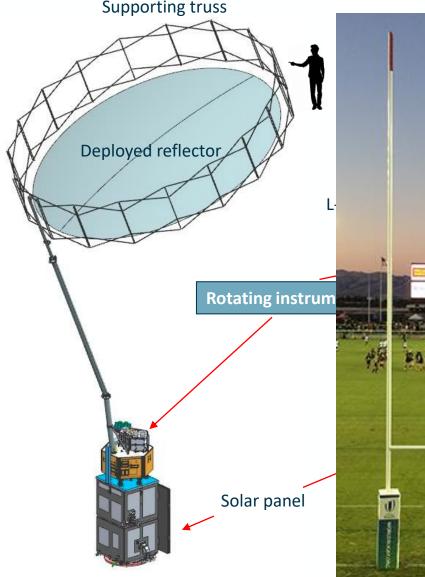
The CIMR Payload













alancing mass

lold down release echanisms

Rotating instrument

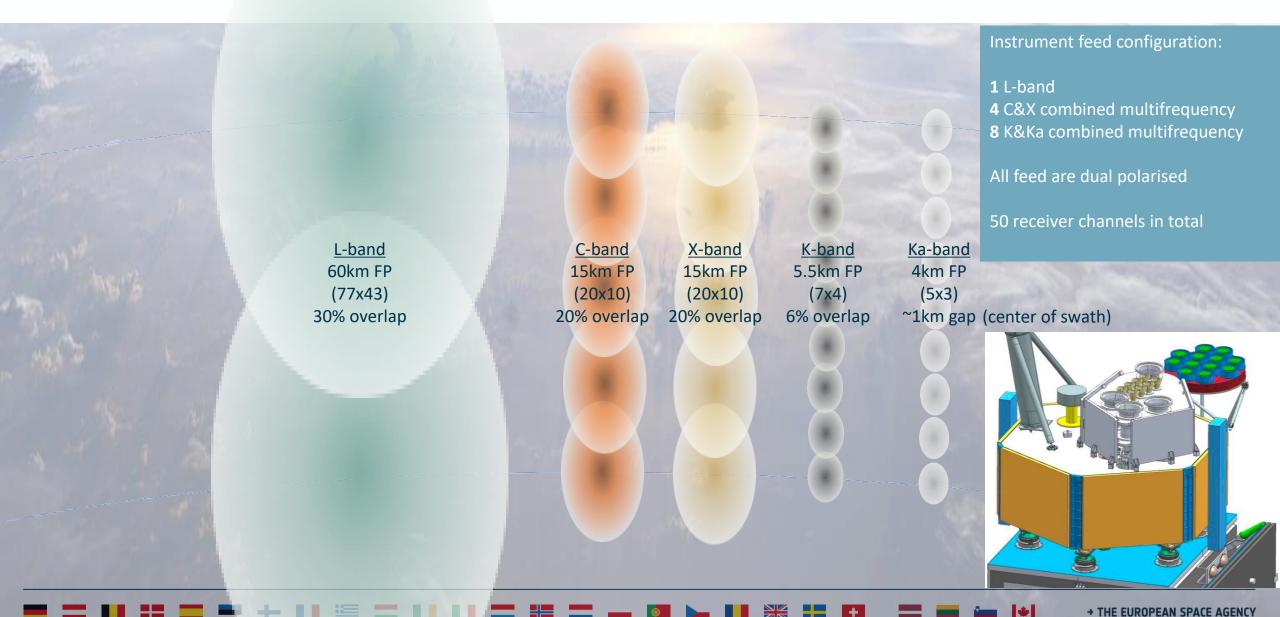
- 50 receiver channels in total (~11GHz total bandwidth), incuding dual linear polarisation.
- Full Modified Stokes parameters provided.
- Each channel uses internal calibration.
 - Hot and Active Cold Load (ACL).
- Detection is done in digital domain.
- All channels have onboard **RFI processor**.
 - To identify interference and remove it from the mesurement.
- All the above done in rotating part of the satellite (due to limitation in data transfer through the rotary joint to the fixed part).

Stowed reflector & boom



The CIMR Measurement Principle Footprint sizes and overlap for all frequencies @ center of swath







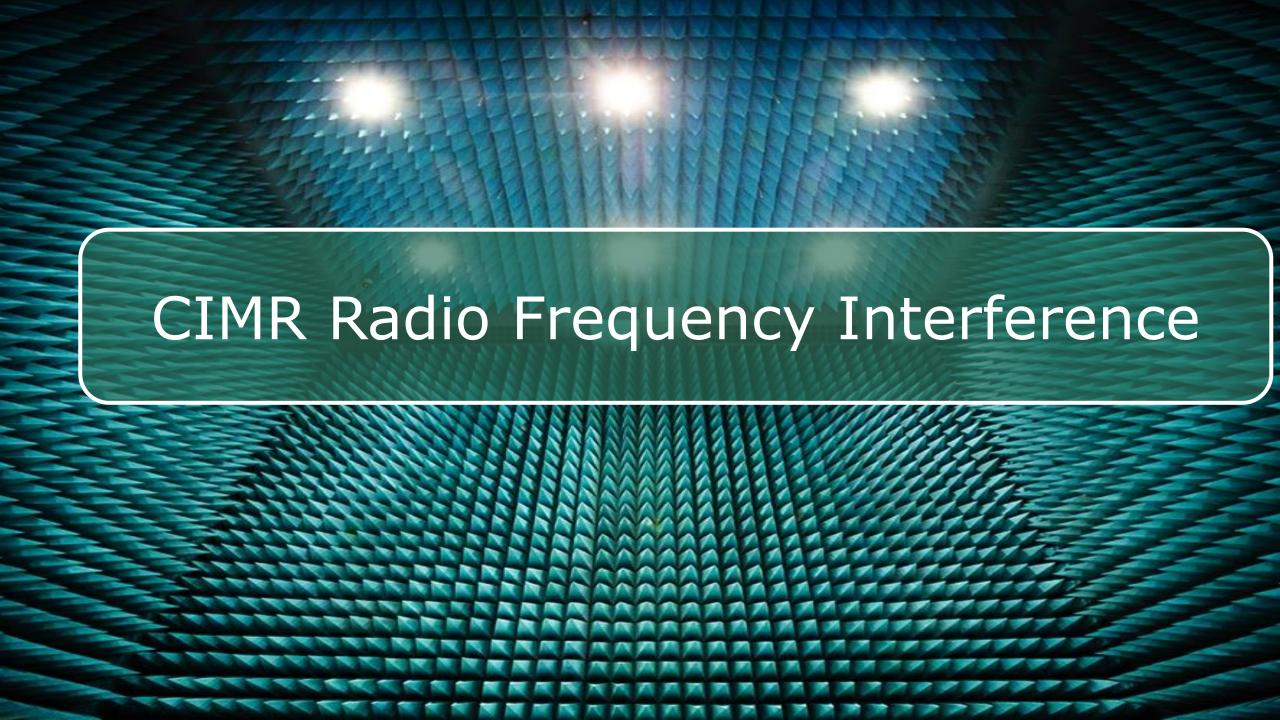
opernicus CIMR Key Radiometric Performances



±1						
Label	Mission Priority	Primary	Primary	Primary	Primary	Primary
ID-080-1-1	Addressing CIMR Objectives	ALL	ALL	ALL	ALL	ALL
ID-080-1-14 (MRD-250)	ITU EESS (passive) allocated band and band centre frequency (MHz)	1.4 – 1.427 1.4135	6.425–7.250 6.8375	10.6-10.7 10.65	18.6-18.8 18.7	36-37 36.5
ID-080-1-2 (MRD-240)	Channel centre frequency ¹⁴ [GHz]	1.4135	6.925	10.65	18.7	36.5
ID-080-1-3 (MRD-380)	Maximum channel bandwidth [MHz]	25	300	100	200	300
ID-080-1-4 (MRD-300)	footprint size [km]	<6016	≤15	≤15	≤5.5	<5 (goal=4)
ID-080-1-5 (MRD-420)	L1b Radiometric resolution [K] NEΔT for zero mean, 1-sigma at 150 K	≤0.3	≤0.2	≤0.3	≤0.4 (goal: ≤0.3)	≤0.7
ID-080-1-6 (MRD-430)	Dynamic Range [K]	<u>Kmin</u> =2.7, <u>Kmax</u> =340				
ID-080-1-7 (MRD-440, MRD-450, MRD-460)	L1b Radiometric Total Standard Uncertainty ¹⁷ [K, zero mean, 1-sigma)]	≤0.5	≤0.5 (goal ≤0.4)	≤0.5 (goal: ≤0.45)	≤0.6 (goal: ≤0.5)	≤0.8
ID-080-1-8 (MRD-560)	Polarisation	Full Stokes (see MRD-550, MRD-560, MRD-570)				
ID-080-1-9 (MRD-170)	Swath width [km]	>1900				
ID-080-1-10 (MRD-270)	Observation Zenith Angle [deg]	55.0 ±1.5				
ID-080-1-11 (MRD-470)	L1b Radiometric stability over lifetime [K, zero mean, 1-sigma]	≤0.2	≤0.2	≤0.2	≤0.2	≤0.2
ID-080-1-12 (MRD-480, MRD-490)	L1b Radiometric stability over orbit [K, zero mean, 1- sigma]	≤0.2	≤0.15 (goal=0.1)	≤0.15 (goal=0.1)	≤0.2	≤0.2
ID-090-1-13 (MRD-660)	L1b geolocation uncertainty [km]	≤1/10 of ID-080-1-4 (see MRD-660)				

The CIMR instrument remains on track to meet these performances

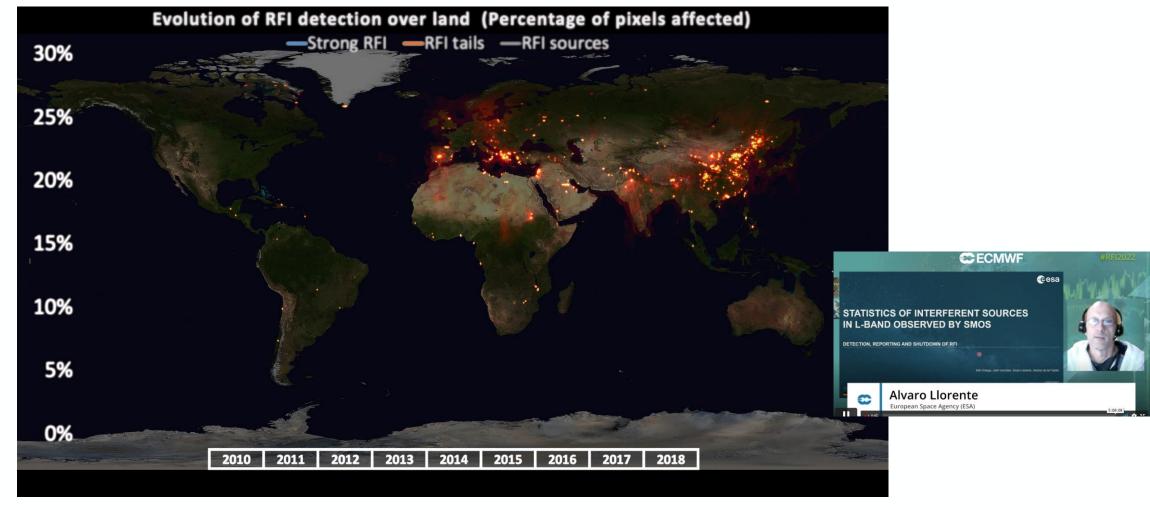
https://esamultimedia.esa.int/docs/EarthObservation/CIMR-MRD-v4.0-20201006_Issued.pdf





pernicus Evolution of RFI in the 1400-1427 MHz band



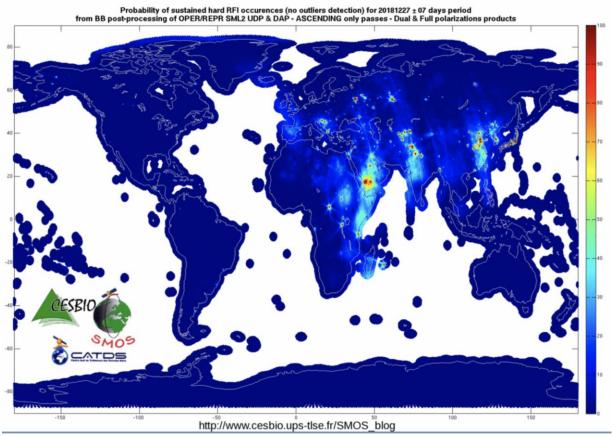


The SMOS experience has been a great lesson learned

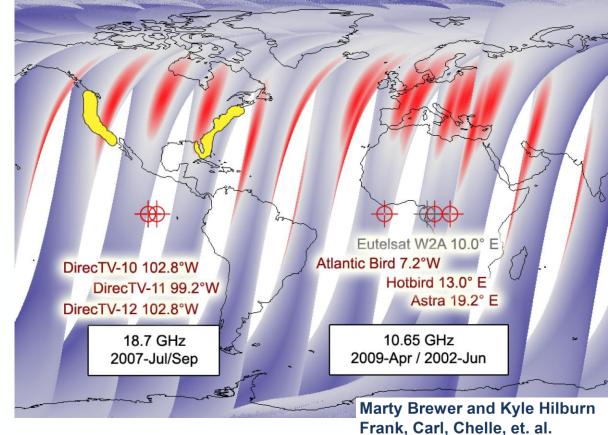


opernicus CIMR: RFI @L-, C-, X- K- and Ka-band !!!





http://www.cesbio.upstlse.fr/SMOS_blog/smos_rfi/?q=image/3-latest-global-15days-asc



Remote Sensing Systems AMSR Science Team Meeting Portland, Oregon, 2012.Sep.11-12



CIMR: RFI @L-, C-, X- K- and Ka-band !!!



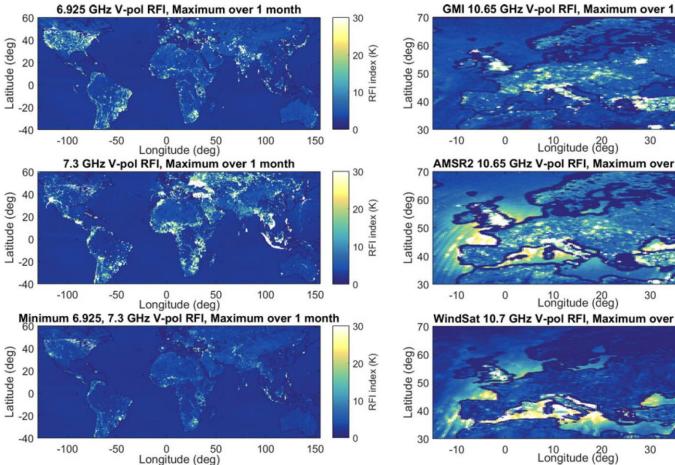


Fig. 7. RFI Mitigation effectiveness for AMSR2 the C-Band Channels. Top: Maximum RFI index over a month for the 6.925 GHz V-pol channel. Middle: Maximum RFI index over a month for the 7.3 GHz V-pol channel. Bottom: For each measurement over the month, the minimum RFI index of the two frequency bands is chosen, and the maximum result over the month is shown, AMSR2 data taken from October 2017.

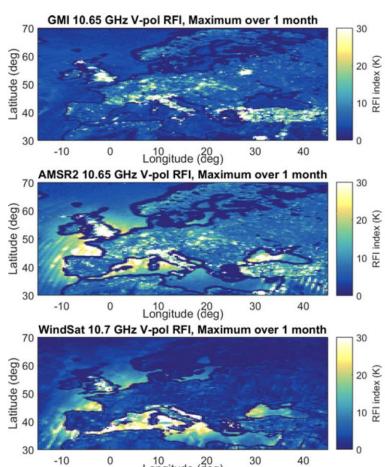


Fig. 8. 10.65–10.7 being in the protected band, GMI (top) does not exhibit reflected RFI around Europe but does pick up substantial RFI over land. AMSR2 (middle) is also in the protected band, but does see reflected RFI. WindSat (Bottom) whose bandpass is in the nonprotected satellite transmission band exhibits significant interference. GMI and AMSR2 data are taken from October 2017 and WindSat data are taken from September 2014.

- RFI is a fundamental issue for CIMR
- It must be addressed holistically if the Mission is to reach specified performance

Radio Frequency Environment for Earth-Observing Passive Microwave Imagers

David W. Draper ¹⁰, Member, IEEE



opernicus Drivers: Radio Frequency Interference (RFI)



5G Could Interfere With Weather Satellites, Scientists Warn

The future comes with some difficulties.

Big issue for C-band and likely X-band



pernicus RFI: ITU issues and approach

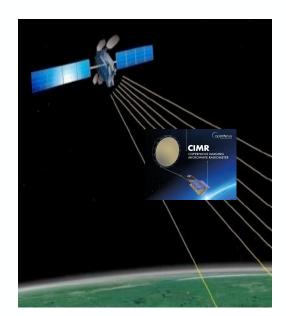


- L-band: Important to have a very well defined channel selectivity (bandpass and receiver) centered in the
 Earth Exploration Satellite Service(EESS) passive band 1400-1427 MHz (SMOS/SMAP experience show RFI
 from strong radars in adjacent bands).
- C-band: *EESS(passive)* is weak and we cannot claim protection (5G issues coming?). AMSR-xx shows significant RFI
- **X-band:** heavily used by the GEO/NGEO Fixed Satellite Service (FSS) on a primary basis and European video links (*5G issues coming?*)
- **K-band:** sharing regulatory constraints of EESS(passive) and the FSS downlinks. FSS (s-E) has allocation in the range 17.3 to 21.2 GHz, allocation to EESS(passive) falls in the middle.
- **Ka-band:** Powerful RADARS (e.g. KREM) operating in the lower adjacent band. can blind, even damage, the receiver. The FSS(downlinks) operate above 37.5 GHz and are target for development of future LEO megaconstellations.
- Solutions:
 - 1. Protect from damaging RFI
 - 2. Detect and mitigate using on-board processors in NRT3 Hours (Channel selectivity is important)
 - 3. Reprocess using on-ground data tools and techniques

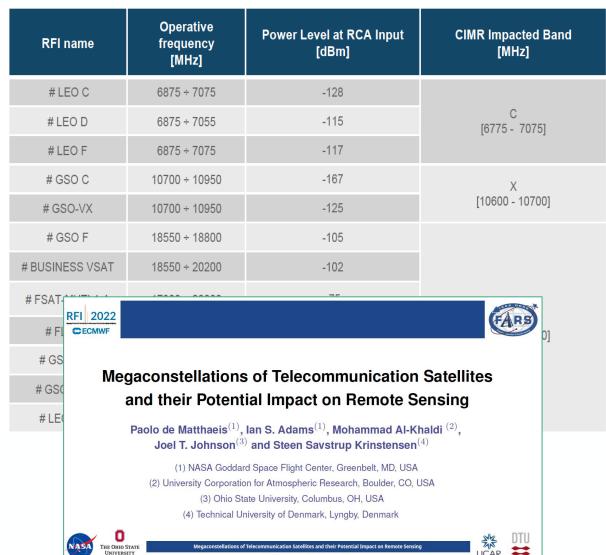


ernicus RFI environment: Satellite behind CIMR ...





Space to Ground transmission through the CIMR reflector



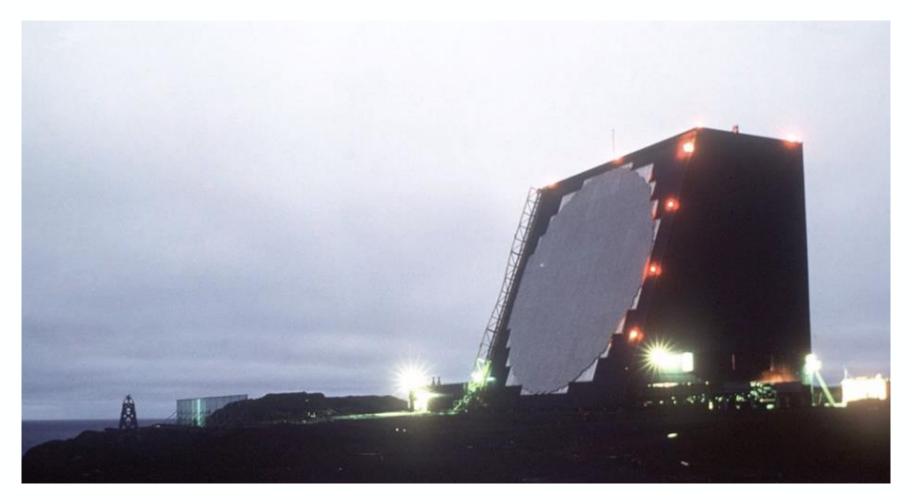
Transparent mesh at High frequencies so we receive RFI from behind the reflector from other satellites

List shows list of satellites that are important

All RFI well attenuated
– below the levels
from ground – but still
to be considered







https://en.wikipedia.org/wiki/Cobra Dane#/media/File:Cobradane.jpg



ernicus RFI environment: Earth to CIMR ...



RFI Name	CIMR Band	Frequency Range [MHz]	Frequency Distance [MHz]	Power Level at RCA Input [dBm]
SMOS RFI	L	-	In Band	-63,92
Radar A	L	1175 ÷ 1400	1	31,63
Radar A	L	-	In Band	22,7
ASR 4	L	1215 ÷ 1400	1	-6,78
FSS GES	С	5725 ÷ 7075	In Band	-64,2
Military Uplink (L)	С	5850 ÷ 6450	325	-24,02
Radar L	С	5350 ÷ 5850	925	1,23
Deep Space Station	С	7145 ÷ 7190	70	31,25
SBX	X	9000 ÷ 10000	600	26,61
Unknown (X)	X	9200 ÷ 10400	200	39,85
Haystack	X	9300 ÷10300	300	26,53
FSS GES	X	10700 ÷ 11700	0	-47,35
FSS GES	Ku	-	In Band	-100,31
BSS GES	Ku	17700 ÷ 18400	200	-63,53
Unknown (Ku)	Ku	17200 ÷ 17300	1300	20,67
FGAN	Ku	15900 ÷ 17500	1100	-5,43
KREMS	Ka	34000 ÷ 36000	350	25,3

- RADAR-A (Cobra Dane) should be filtered. This radar type may operate up to 1400MHz (edge of CIMR Lband channel), and even inside the CIMR band outside of Europe
- For the L-band channel, the operational characteristics of Radar A means that a limiter diode needs to be inserted in the receiver chain prior to the first amplification chain
- For C, X, K and Ka band, filters will be employed to protect the receiver channels from strong out of band RFI sources
- Note, C-band is TBD. A limiter diode may be used intead/in combination with filters in order to optimise mass/losses prior to the first ampification stage in order to optimise sensitivity of the receiver chain.
- But...Filters add mass and loss so we may consider limiters as a mass saving solution (TBD)
- KREMS should be **filtered** as we have some separation

At the Receiver we can withstand ~13 dBm for L- and C-band

RFI Environment – How CIMR protects itself



IF RFI exists, THEN perforance of CIMR is degraded:

- Potential physical damage to the instrument (end of mission...)
- Data loss (hopeless numbers)
- Increase in noise (NeDT → poor sensitivity important for SST and Salinity, must be managed well if RFI is mitigated)
- Incorrect retrieval of geophysical parameters (if undetected and unflagged RFI)
- CIMR is threatened by 3 key issues:
 - 1) Protection from strong RFI sources (ground or space sources)
 - 2) Detection and mitigation of in-band RFI sources from ground
 - 3) Detection and mitigation of in-band RFI sources from space (other satellites)
- CIMR is designed to provide products within 3 hours of observation and thus an on-board RFI processor is required to minimise the impact of RFI:
 - What can we salvage in NRT3H? What can we do on ground (what do we send to ground?)

opernicus CIMR on-board RFI mitigation approach 19:20 - 19:40 **SIMR** channel will have For in-band inter ation processors wi dedicated RFI/ RFI Detection and Mitigation Processor for the range of algo Copernicus Imaging Microwave Radiom tor the Copernicus Imaging Microwave Radiom tor the Krictoneon /Torhnical I Inivarcit glitch, polar/ Speaker: Steen Savstrup Kristensen (Technical University of CIMR architecture design algorithms and more/new algo-Frequency Sub-banding to allow identification and remain for the Next Generation Satellite Radiometers and reconstruction of a useful measurement: Incre **∲IEEE** in data rate Note: When RFI is mitigated by removing a sub-band vstrup Kristensen ; Sten Schmidl **NeDT** increases. 020m **Speir Application to Future** Additional tests using **Kurtosis requires computation** European Space of Full Stokes Publisher: IEEE Steen S. Kristensen; Niels Skou; Sten S. Søbjærg When RFI is detected the intent is to **send as much information to** ground as feasible (limit is the throughput of our rotating between the instrument and spacecraft and X-band downlink).



CIMR: 3 on-board RFI NRT detection and mitigation modes



- Normal mode: Operation Mode: RFI detection and mitigation on-board using DTU approach (next talk).
 - Output: Native original data and RFI mitigated data sent to ground + number of removed time-frequency matrix cells as an indicator of # sub-bands removed
 - But don't know which ones have been mitigated so we have challenges to reconstruct NeDT
- <u>Diagnostic mode</u>: initiated by T/C (configurable for 1 feed horn of one frequency both H&V):
 - Continuously send_full_RFI_time_frequency_matrix_to_ground
 - Limited data take at this time...
- On-Event Mode (OEM): initiated by T/C (configurable for 1 feed horn of one frequency, both H&V):
 - IF (RFI) THEN send_full_RFI_time_frequency_matrix_to_ground

Final configuration and data type/flags/quantities (TBC) all being studied now in Phase B2.

CIMR Band	L	С	X	Ku	Ka
Accumulation Time [ms]	0,545	0,7	0,685	0,504	0,736
Rate [Hz]	1835	1429	1460	1984	1359
#Sub-Bands	18	215	72	143	215
#Time Bin	20	4	4	2	1
#Feed	1	4	4	8	8
#Parameters	6	6	6	6	6
Feed Id [bits]	8	8	8	8	8
Parameters Data Size [bits]	16	16	16	16	16
T-F bin Id [bits]	16		16	16	16
Spare [bits]	16	16	16	16	16
Single Feed Data Rate [Mbps]	3,96	36,87	12,61	34,06	35,06
Band Data Rate [Mbps]	3,96	147,47	50,46	272,50	280,50
Total Data Rate [Mbps]	754,89				

Data rate is obtained from Time-Frequency Matrix division:
Accumulation time: time length of a time bin
#Sub-bands: number of sub-bands of in a Band

#Sub-bands: number of sub-bands of in a Band

Transferred data must include also Kurtosis parameter (KH, KV)



ernicus Ideal scenario is On Event Mode



- The On-Event Mode could be triggered by the detection of an RFI in flight (assumes well calibrated algorithms):
 - When RFI is detected, all sub-bands related to the relevant time bin are downloaded;
 - We have data to reconstruct NeDT securely;
 - We can monitor the evolution of RFI
 - We can decide if/when to update on-board RFI algorithm parameters (changing RFI environment) based on data;
 - We have evidence of RFI for Regulatory compliance tasks
 - We can reprocess data on ground using updated RFI processing algorithms
- Currently studying how to optimize transfer of data across the CIMR rotating joint further
 - In principle, the RFI Event can involve more than one Time
 Bin how likely is this in general across all channels?
 - We have data rate limitations across the CIMR rotating joint...

The number of 'events' that can be handled assuming that only 1 feed of 1 band is involved in the detection with current data rates

CIMR Band	Single Feed Data Size [kbit]	Maximum No of events per band per second	
L	1,73	6019	
С	20,64	504	
X	6,91	1505	
Ku	13,73	758	
Ka	20,64	504	

Scientific Data computed from on-board processors includes four measurements in all time-frequency matrix bins:

- PH, Power of H channel;
- PV, Power of V channel;
- P3, 3rd Stokes parameter;
- P4, 4th Stokes parameter;





→ THE EUROPEAN SPACE AGENCY



opernicus Current status - the path to reality



Thales Alenia Italy Italy signed contract to Prime CIMR mission Phase B2/C/D development (13/11/2020)

Preliminary Design Review (2022)

Mission Requirements Document available at https://esamultimedia.esa.int/docs/EarthObservation/CIMR-MRD-v4.0-20201006_Issued.pdf

Launch of CIMR-A in 2028+ (CIMR-B few years later)

RFI remains a major challenge for CIMR

We are addressing RFI issues as a core design element









Thank you Any Questions?

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European Space Agency